UNCLASSIFIED

20000920147

REPORT DOCUMENTATION PAGE	READ INSTRUCTIONS
1. REPORT NUMBER 2. GOVT ACCESSION NO	BEFORE COMPLETING FORM 1. RECIPIENT'S CATALOG NUMBER
HDL-TM-82-10 ADAI214	30
4. TITLE (and Subtitio)	S. TYPE OF REPORT & PERIOD COVERED
Fluerics 42: Some Commonly Used Laminar Fluidic Gain Blocks	Technical Memorandum
, razaro darii brooko	S. PERFORMING ORG. REPORT NUMBER
7. AUTHOR(e)	9. CONTRACT OR GRANT NUMBER(#)
Tadeusz M. Drzewiecki	PRON: 1F2E0001011FA9
 Performing organization name and adoress U.S. Army Materiel Development and 	10. PROGRAM ELEMENT, PROJECT, TASK AREA & WORK UNIT NUMBERS
Readiness Command Alexandria VA 22333	Program Ele: 6.11.02.A
11. CONTROLLING OFFICE NAME AND ADDRESS	12. REPORT DATE
Harry Diamond Laboratories 2800 Powder Mill Road	September 1982
Adelphi, MD 20783	13. HUMBER OF PAGES
14. MONITORING AGENCY NAME & ADDRESS(II dillerent from Controlling Office)	18. SECURITY CLASS. (of this report) UNCLASSIFIED
	154. DECLASSIFICATION/DOWNGRADING SCHEDULE
18. DISTRIBUTION STATEMENT (of this Report)	
Approved for public release; distribution	on unlimited.
17. DISTRIBUTION STATEMENT (of the abetract entered in Block 20, if different fro	om Report)
•	·
18. SUPPLEMENTARY NOTES	
HDL Project: A41234 DRCMS Code: 611102.H.440011	·
DA: 1L161102AH44	ł
19. KEY WORDS (Continue on reverse side if necessary and identify by block number, Fluidics Pressu	re regulators
Fluerics Laminar	jet angular rate sensor
Laminar proportional amplifier Finebla	
	odischarge machining
Servolvalve Photoch	nemical etching
This report presents data and o information on many commonly used lamin	perating experience
two to eight stages. In addition, as	an aid to design, a 📑
short computer program is presented, sui	table for use with a
pocket programmable calculator. Output	s from this program
are individual stage data including nomi and flows and staged gain. Also availa	
and the bandwidth at 90 deg of phase	se shift. Several

DO 1 JAN 79 1473 EDITION OF 1 NOV 65 IS OBSOLETE

UNCLASSIFIED

1 SECURITY CLASSIFICATION OF THIS PAGE (

Reproduced From Best Available Copy

SECURITY CLASSIFICATION OF THIS PAGE(When Date Entered)

20. Abstract (cont'd)

examples of this program are given to cover multiple-stage gain blocks. As an example of the utility of the program, a step-by-step tradeoff study is presented for the Massachusetts Institute of Technology fluidic servovalve, for which an attempt is made to maximize bandwidth and minimize leakage flow.

UNCLASSIFIED

2 SECURITY CLASSIFICATION OF THIS PAGE(Phon Date Entered)

CONTENTS

		,			•		Page
1.	INTR	ODUCTIO	Эн	• • • • • • • • • •	• • • • • • • • • •	• • • • • • • • • • • • •	5
2.	SIMP	LE STAC	GING DESIGN	rrogram			6
3.	•						
٥.	COMM						
	3.1	Two-St	age Gain B	locks	• • • • • • • • • •	• • • • • • • • • • •	10
		3.1.1					
		3.1.2	Two-Stage	Moderately	Pressure-St	aged Gain Bl	ock11
	3.2	Three-	Stage Gain	Blocks	• • • • • • • •	• • • • • • • • • • •	13
		3.2.1	Three-Sta	ge Self-Stag	ed Gain Blo	cks	13
		3.2.2	Common Su	oply Pressur	e Gain Bloc	ks	15
		3.2.3					
		3.2.4	Power- or	Flow-Gain G	ain Blocks.	• • • • • • • • • • • • • • • • • • • •	24
	3.3	Four-S	stage Gain 1	Blocks			25
	3,4					s System	
	3.5						
4.	DISC	ussion	AND CONCLUS	SIONS			28
እር _ጉ	MOMT.N	DOWNENT	·S				29
LIT	ERATU	RE CITE	D	• • • • • • • • • • •	• • • • • • • • • •	• • • • • • • • • • • • • • • • • • • •	,31
ois	TRIBU	TION	• • • • • • • • •		•••••	• • • • • • • • • • •	53
			•	APPEND	ICES		
١	-BASI	C PROGR	AM AND REP	RESENTATIVE	EXAMPLES		33
3	-REPR	ESENTAT	IVE STACKI	G ORDERS		• • • • • • • • • • •	47
				FIGU	RES		
•						ar proportion	
2.	Two	-stage :	self-staged	gain block	:	• • • • • • • • • • •	10

FIGURES (Cont'd)

		Pag
3.	Fluidic capillary pyrometer gain bloc	:k:12
4.	Three-stage self-staged gain block:	
5.	Three-stage common supply pressure ga	in block:16
6.	Three-stage, parallel-element, common gain block:	
7.	Three-stage maximum dynamic range gai	n block:19
8.	Three-stage migh-input-impedance gain	block:19
9.	Three-stage preamplifier for laminar	jet angular rate sensors:20
10.	Bode plot of flueric servovalve three	-stage gain block21
11.	Four-stage pressure regulator gain bl	.ock:26
12.	Design program output for Harry Diamo public address system	
	TABLES	
1.	Optimization of Fluidic Servovalve fo Output-to-Input Resistance Ratio, and	
2.	Makeup of Eight-Stage Gain Block for Stabilization System	
		Accession For
	DTIC FILECTE HOV 1 6 1982	NTIS GRAWI DTIC TAB Uncount mond Justifies from
7	В	Avails to Cod.s Avail and/or Dist Special

1. INTRODUCTION

In developing fluidic systems, it is most often necessary to provide a certain amount of gain. Whether it is to boost sensor outputs to useful levels or to provide high gain feedback, each application requires amplification of differential pressures. Since the gain of a single standard laminar proportional amplifier (LPA) is about 10, it is necessary to stage LPA's if more gain is desired.

Staying LPA's has been dealt with extensively in the literature; I^{-6} however, performance of preferred topologies has not in general been discussed. Certain advantages, not specifically obvious, lie with certain specific designs.

The most commonly used gain block is the self-staged gain block. Although one of the most important operational advantages of such a device is that the high single-stage LPA input-to-output resistance ratio is maintained, perhaps the most compelling reason for its use is the ease with which assembly and manifolding of supply pressure occur since only one supply pressure is needed. Other topologies, however, use a single common supply pressure to each amplifier, and most gain blocks use a single supply which is then manifolded down. Therefore, this report presents experimental data for many different gain blocks. Although operational characteristics are stressed, a design rationale where it is not obvious is presented. A simple program, developed for use with a pocket programmable calculator, provides the basis for the design of a gain block and is treated first.

¹F. M. Manion and G. Mon, Fluerics 33: Design and Staging of Laminar Proportional Amplifiers, Harry Diamond Laboratories, HDL-TR-1608 (September 1972).

²F. M. Manion and T. M. Drzewiecki, Analytical Design of Laminar Proportional Amplifiers, Proceedings of HDL Fluidic State-of-the-Art Symposium, 1 (October 1974).

³T. M. Drzewiecki, D. Wormley, and F. M. Manion, A Computer-Aided Design Procedure for Laminar Fluidic Systems, J. Dyn. Sys. Meas. Control, 97, Series G, No. 4 (December 1975).

^{*}C. Mon, Flueric Laminar Gain Blocks and an Operational Amplifier Scaler, Harry Diamond Laboratories, HDL-TR-1730 (December 1975).

⁵T. M. Drzewiecki, A Fluidic Audio Intercom, Proceedings of 20th Anniversary of Fluidics Symposium, American Society of Mechanical Engineers special publication G00177, Chicago, IL (November 1980).

⁶M. Cycon and D. Shaffer, Design Guide for Laminar Flow Fluidic Amplifiers and Sensors, contract with Garrett Pneumatic Systems Division, Harry Diamond Laboratories, HDL-CR-82-286-1 (March 1982).

2. SIMPLE STAGING DESIGN PROGRAM

The in-stage gain of a standard LPA is determined by the operating bias pressure and the load that the LPA is driving. The sensitivity of gain to bias pressure is given by the relationship?

$$G_{\text{biased}}/G_{\text{zero bias}} = 1/[1 + 0.6(P_{\text{bias}}/P_{\text{g}})]$$
, (1)

which is shown in figure 1 for some typical amplifiers.

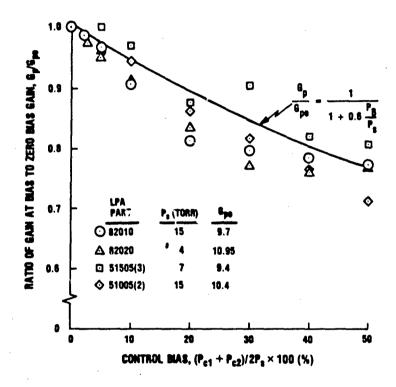


Figure 1. Gain as function of bias for standard laminar proportional amplifier of Harry Diamond Laboratories.

If one assumes that under most leading conditions the recovered pressure of a preceding stage is at least 33 percent (for aspect ratio, $\sigma > 0.7$), then the expression for gain can be approximated in terms of the preceding (n - 1) stage supply pressure, so that

$$G_{biased}/G_{zero\ bias} = 1/[1 + 0.2(P_{s(n-1)}/P_{sn})]$$
 (2)

⁷T. N. Drzewiecki, Fluerics 38: A Computer-Aided Design Analysis for the Static and Dynamic Port Characteristics of Laminar Proportional Amplifiers, Harry Diamond Laboratories, HDL-TR-1758 (June 1976).

The gain of an LPA is also determined by the load into which the amplifier is operating so that

$$G_{load}/G_{blocked} = 1/[1 + (R_o/R_L)]$$
, (3)

where $R_{\rm O}$ is the LPA output resistance and $R_{\rm L}$ is the load resistance. Since the output resistance for almost any case is 5 $R_{\rm ON} = 0.5 P_{\rm SN}/Q_{\rm SN}$ and the in-stage load is the input resistance of a succeeding (n + 1) stage, 5 then

$$R_{L} = R_{i(n+1)} = 0.75P_{s(n+1)}/Q_{s(n+1)}$$
.

In such a manner, equation (3) becomes

$$G_{load}/G_{blocked} = 1/[1 + 0.667(P_{sn}/P_{s(n+1)})(Q_{s(n+1)}/Q_{sn})]$$
 (4)

The phase shift for small phase angles is determined by the signal transport lag in the device. The total gain block phase shift, as a minimum, is the sum of the phase shifts of each stage:

$$\phi_n = 360 \text{ rf} \tag{5}$$

and

$$\phi = \sum_{n=1}^{N} \phi_n \tag{5a}$$

so that

$$\phi = 360f \sum_{n=1}^{N} \tau_n , \qquad (5b)$$

where ϕ is in degrees, τ is the signal transport time across an LPA (equal to twice the particle transport time), ⁵ and f is the frequency in hertz. The signal transport time is

⁵T. M. Drzewiecki, A Fluidic Audio Intercom, Proceedings of 20th Anniversary of Fluidics Symposium, American Society of Mechanical Engineers special publication G00177, Chicago, IL (November 1980).

$$\tau = 2x_{sp}/c_d(2P_s/\rho)^{1/2}$$
 (6)

Since in the standard LPA the nozzle-to-splitter distance, $x_{\rm sp}$, is eight nozzle widths (8bs), τ becomes

$$t = 16b_{\rm s}/c_{\rm d}(2P_{\rm s}/\rho)^{1/2}$$
 (7)

To determine the frequency at which total phase shift is ϕ degrees, one uses equations (5b) and (7) to get

$$f = (\phi/360) / \left[16 \sum_{n=1}^{N} b_{sn}/c_d (2P_{sn}/\rho)^{1/2} \right] ,$$
 (8)

where c_d is the LPA discharge coefficient (c_d = 0.7 for the nominal operating point) and ρ is the fluid density.

Equations (1), (3), and (8) can be written in a program for N stages. Just such a program, written in BASIC, is shown in appendix A. The bulk of the program is taken up by setting up the arrays for all the input data. Flow consumption is based on a nominal 0.3 liters/min (LPM) for an LPA with $b_{\rm S}=0.5$ mm at a modified Reynolds number, $N_{\rm R}=120$.

$$N_R^1 = N_R / [2(1 + 1/\sigma)^2]$$

and

$$N_{R} = b_{s} (2P_{s}/\rho)^{1/2}/\nu ,$$

in air,

$$N_{R} = 1000b_{s} \text{ (mm) } [P_{s} \text{ (torr)}]^{1/2}$$
.

Sample printouts are shown in appendix A for nine runs.

Verification of the validity of the results of this program is left to the following sections, in which they are used as a design guide.

3. COMMONLY USED GAIN BLOCKS

In trying to present the information logically, it is easiest to use the number of stages as the primary delineating factor. What follows, therefore, is a compendium of gain blocks and their performance, starting with two stages and ending with eight. Stacking orders for some of the more commonly used circuits are given for standard Harry Diamond Laboratories (HDL) integrated format parts in appendix B.

As a rule, the nominal gain of a single standard LPA is about 10 at $\sigma N_R = 1000$. This value, however, varies depending on the manufacturing process used, the accuracy to which the nominal dimensions are kept, and the aspect ratio. As a result, a normal span of gain variation is from 7 to 12. As the aspect ratio gets below 1, the gain decreases from a nominal 10 to as low as 7 at an aspect ratio of 1/3.

The currently accepted methods for fabrication are photochemical etching, fineblanking, and wire electrodischarge machining. ^{8,9} These manufacturing methods are designated by numerical prefixes to the four-digit designation numbers as 5, 6, and 8, respectively, forming a five-digit LPA part number. The two digits immediately following the manufacturing prefix correspond to the nozzle width in mils, and the last two digits are the lamination thickness in mils. Alphanumeric suffixes in brackets may designate special dimensions or functions; for example, [R] refers to a rectifier. In this fashion, a fineblanked LPA with $b_{\rm g} = 0.5$ mm (0.020 in.) and $\sigma = 0.75$ is designated as P.N. 62015, and an etched LPA with $b_{\rm g} = 0.25$ mm (0.010 in.) and $\sigma = 0.5$ is designated as P.N. 51005.

In the following sections describing gain blocks, the average value of a single-stage blocked-pressure gain is given to provide a yardstick against which to evaluate possible alternative performances. The reader should remember that net gain is the product of the blocked single-stage gain raised to the power N (where N is the number of stages) multiplied by the net losses. The losses generally remain the same, independent of the blocked gain, so that if one is interested in the performance of a gain block made up of LPA's of gain different from those reported here, one merely multiplies by the ratio of new gain to old gain raised to the Nth power. For example, a three-stage gain block is reported here to have a gain of 200 using LPA's with a blocked single-stage gain of 8, but is reproduced by using components with a gain of 10. The new gain block has a gain of $(10/8)^3$ higher, or 391.

⁸L. Scheer, J. Roundy, and J. Joyce, Manufacturing Techniques for Producing High Quality Fluidic Laminates in Production Quantities, Proceedings of 20th Anniversary of Fluidics Symposium, American Society of Mechanical Engineers special publication G00177, Chicago, IL (November 1980).

⁹R. M. Phillippi, A Study of Fineblanking for the Manufacture of Flueric Laminar Froportional Amplifiers, Harry Diamond Laboratories, HDL-TM-77-8 (May 1977).

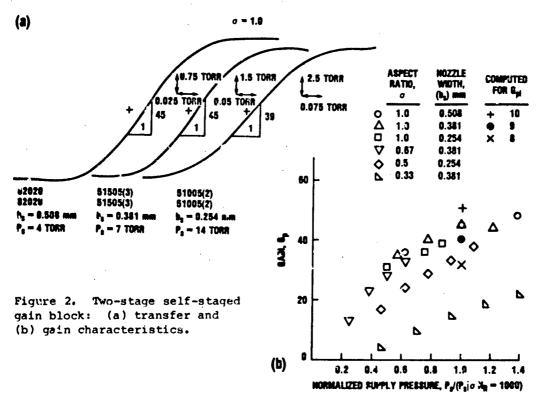
医乳毒素 地名加加克利克 医多种性 医多种性性 医多种性性 医克拉特氏 医克拉特氏病 医克拉克氏病

3.1 Two-Stage Gain Blocks

Several two-stage gain blocks have been implemented at HDL, but two are most common. The first is a self-staged gain block, and the second is a moderately pressure-staged gain block that is extensively used in fluidic capillary pyrometer circuits. 10

3.1.1 Two-Stage Self-Staged Gain Block

Figure 2 shows a collection of data for various size and aspect ratio two-stage self-staged gain blocks. Representative transfer characteristics are given at the nominal operating point in figure 2(a), and the gain as a function of normalized supply pressure is given in figure 2(b). Also shown in figure 2(b) are the computed values of gain using the program given in appendix A. As may be observed, there is good correlation. A generic stacking order is given in appendix B.



¹⁰R. M. Phillipri, T. M. Drzewiecki, T. Negas, and H. S. Parker, Design of a Fluidic Capillary Pyrometer for Contact Duty at Temperatures to 2750°C, Proceedings of 6th International Symposium on Temperature, Washington, DC (15 to 19 March 1982).

3.1.2 Two-Stage Moderately Pressure-Staged Gain Block

To get more gain using low-aspect-ratio, small-size devices with high impedance and high output pressure, some slight amount of pressure staging is obtained by cascading a device with an aspect ratio of 0.33 with one of 0.67 for the first stage. These stages operate at different pressures, so to operate the device from the common last-stage pressure, a dropping resistor must be employed.

The computer program generates values for nominal supply pressure and flow so that values of dropping resistors can be estimated by taking the difference between last-stage pressure and desired-stage pressure and dividing by the desired-stage supply flow. In this manner,

$$R_{d} = (P_{N} - P_{i})/Q_{i} ,$$

where P_N is the common supply pressure (pressure to the last stage), P_i is the pressure to the stage of interest, and Q_i is the flow to the stage of interest.

For the case at hand, P_1 = 16 torr, P_2 = 64 torr, and Q_1 = 0.225 LPM (see example A-1 in app A). This requires a resistance of about 200 torr/LPM between the common supply and the first-stage supply. Standard available resistances, at a nominal flow rate of 0.05 LPM, are 35, 200, 300, and 500 torr/LPM. These correspond to P.N. 5221a nozzles with b_8 = 0.51 mm and σ = 0.5, b_8 = 0.56 mm and σ = 0.23, b_8 = 0.51 mm and σ = 0.25, and b_8 = 0.51 mm and σ = 0.20.

The 35 and 300 resistors are available in fineblanked parts, and the 200, 300, and 500 resistors are available etched. For flow rates above 0.05 LPM, the orifice term adds resistance so that a nominal 200-torr resistor at 0.225 LPM has too high a resistance for this job. By using two 300-torr ref stors in parallel at a flow rate of 0.112 LPM each, the net 150 torr/LPM is increased so that it is close to 200. (It is good practice always to use dropping resistors in parallel to cut supply noise. Keeping the flow to less than 0.05 LPM per resistor further reduces noise.)

Figure 3(a) shows typical transfer characteristics and figure 3(b) shows the gain versus the supply pressure for this fluidic capillary pyrometer gain block using fineblanked parts of nominal gain just under 9. The computed value of 61, shown in appendix B using a single-stage gain of 8.5, is in good agreement with the value of 62 actually obtained.

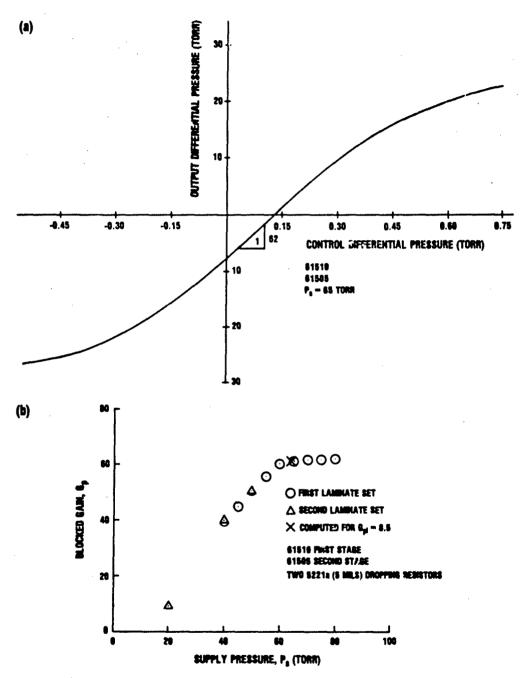


Figure 3. Fluidic capillary pyrometer gain block: (a) transfer and (b) gain characteristics.

The stacking order for standard C format assembly is given in appendix $\ensuremath{\mathtt{B}}_\bullet$

3.2 Three-Stage Gain Blocks

Perhaps the most commonly used generic gain block is composed of three stages. Gains from 100 to 1000 are readily obtainable with frequency response well past the normal AM radio/telephone range (>3 kHz). Typical use involves reducing the noise floor of commercial pressure transducers or driving of pressure-controlled oscillators. For example, a typical ± 1 psi (± 50 torr) pressure transducer with a signal-to-noise ratio of 2000 can discriminate 0.001 psi (0.05 torr). (This corresponds to 5 mV of noise for a 10-V output.) Typical threshold signals of fluidic sensors are two to three orders of magnitude below 0.05 torr. A no-noise gain block with a gain of 500 boosts the signals above the equivalent electronic noise floor, so that the transducer can effectively discriminate signals of 2×10^{-6} psi (10^{-4} torr).

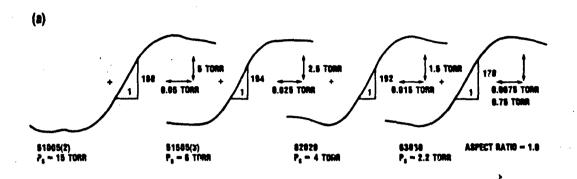
As with two-stage devices, the most common device is the self-staged gain block. Other topologies that follow are pressure or flow staged to one extent or another.

3.2.1 Three-Stage Self-Staged Gain Blocks

The reference gain block is composed of three stages of LPA's with $\sigma=1$. Since LPA's come in standard sizes— $b_s=0.75$, 0.5, 0.375, and 0.25 mm (corresponding to 0.03, 0.02, 0.015, and 0.01 in.)—the performance of the three most commonly used sizes is presented in figure 4a at nominal operating conditions.

Figure 4b shows the variation of gain with supply pressure and computed values from the program. Agreement is good.

The standard components used comprise C format amplifiers 63030, 82020, three stacked 51505's, and two stacked 51005's. The stacking order is similar to that given in appendix B for the two-stage device. There are several guidelines to follow in the assembly of three and more stage gain blocks. The last stage (highest pressure if not self-staged) should be closest to the supply pressure source to minimize high-pressure leaks. The supply flow should be manifolded in such a way that the last stages are fed first with the first stage receiving only that flow which it needs. More succinctly, it is not desirable to have excess flow rushing past the first stage, where it can generate noise that can be amplified through the stack. Furthermore, the first stage vents should always be isolated or decoupled from the rest of the stack, if possible, to prevent spurious latter-stage vent flows from producing noise that can be amplified up the stack (example B-1, use of 5022a plate).



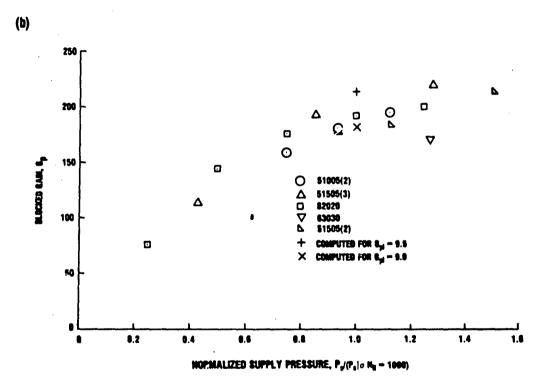


Figure 4. Three-stage self-staged gain block: (a) transfer and (b) gain characteristics.

It is possible to build self-staged gain blocks of different aspect ratios; the gain stays about the same, but the operating pressure changes. For example, a 63020 gain block ($\sigma=0.67$) has a supply pressure of 2.5 torr and a measured gain of 255 with a single-stage normal gain of 11.1.

3.2.2 Common Supply Pressure Gain Blocks

The tremendous advantage of the self-staged gain block is not having to drop supply pressures. This advantage extends to another class of topologies. Since LPA's of the same height, no matter what the nozzle width, operate at the same supply pressure, a moderate increase in staged gain can be achieved by increasing succeeding stage input impedance by decreasing plan view size. In this manner, the four LPA's of standard nozzle width can be staged in various combinations of three or more to get more gain than in the self-staged case. Two such devices are the 63020-62020-51005(4)* and the 62015-51505(3)-51005(3) gain blocks. Figure 5 shows typical transfer and gain characteristics.

Instead of decreasing size, to increase input impedance at constant supply pressure, one may decrease the number of parallel elements per stage and achieve a similar increase in gain. Although this method has the added advantage of requiring only one plan view size amplifier, the cost is a considerable increase in flow consumption. An easy configuration to examine is a first stage with four parallel elements, the second with two, and the third with only one. In this gain block, the input impedance of each succeeding stage is doubled. Figure 6 shows the gain characteristics of this device.

The gain of these blocks, which varies from 300 to 400, is considerably higher than the nominal 200 to 250 gain that the self-staged block exhibits. However, a penalty must be paid for this gain increase in an increased output-to-input impedance ratio. In increasing the output-to-input impedance ratio by paralleling early stages, an additional penalty in flow is exacted for using amplifiers that are all the same size. However, a side benefit of this scheme may be the dramatic reduction in null offset due to cancellation in the multiple elements of the first stage, especially when an even number of parallel

^{*}Metal etched laminates usually come in either 0.1-mm (0.004-in.) or 0.125-mm (0.005-in.) thickness. Since the smaller amplifiers are not available in fineblanked single laminates, the number of laminates used is given in parentheses.

[†]Designation of the number of parallel elements appears as a numerical prefix in brackets to the part number. Hence, four parallel elements of the metal etched parts with $b_{\rm S}$ = 0.254 mm and σ = 1 is designated as [4]51005(2).

stages is employed. Care must be taken to decouple the power supplies in the parallel stages to eliminate power-supply coupling and cross talk that promotes early transition to turbulence. This decoupling moderately increases the manifolding complexity, but still is easier to implement than using dropping resistors.

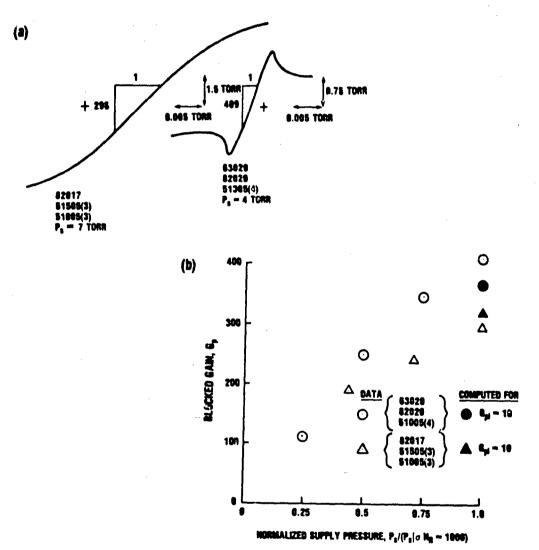


Figure 5. Three-stage common supply pressure gain block: (a) transfer and (b) gain characteristics.

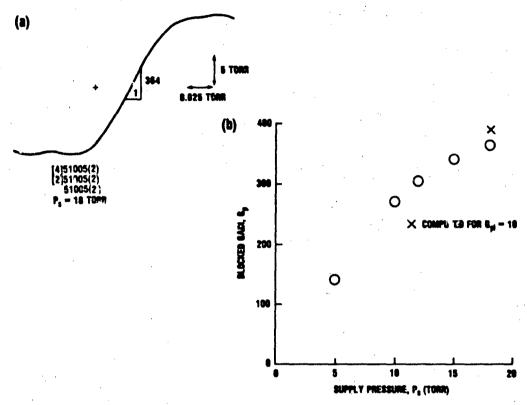


Figure 6. Three-stage, parallel-element, common supply pressure gain block: (a) transfer and (b) gain characteristics.

3.2.3 Pressure-Staged Gain Blocks

Drzewiecki⁵ derives the staging conditions for maximum staged gain and maximum dynamic range as

$$(\sigma_n/\sigma_{n-1})^2(b_{sn}/b_{sn-1}) = 1/G_{pn(staged)} . (9)$$

⁵T. M. Drzewiecki, A Fluidic Audio Intercom, Proceedings of 20th Anniversary of Fluidics Symposium, American Society of Mechanical Engineers special publication G00177, Chicago, IL (November 1980).

Since under optimum loading (almost blocked) the staged gain is nominally 10, then for the same size plan view amplifiers the aspect ratio must decrease by approximately three in each stage to produce maximum gain and dynamic range. When dynamic range is maximized, the jets all deflect the same amount so that the null offset of the gain block is the same percentage as that of the first stage alone. In other words, there is no deflection angle gain in the block. In the self-staged gain block, there is a deflection angle increase; hence, the more stages used, the worse the null offset.

Maximum dynamic range gain block.--By using the lowest aspect ratio to start with for the final stage, one can work backward and determine the aspect ratios for the earlier stages. Normally, $\sigma=0.4$ is the lowest available for $b_g>0.5$ mm (neglecting single-laminate metal etched parts with $\sigma=0.2$ and 0.25). The next-to-last-stage aspect ratio should then be $\sigma=1.2$, and the one before that should be 3.6. With further increase in σ , operating pressure becomes ridiculously low (such as $\sigma=10.8$ and $P_g=0.03$ torr). Therefore, the practical limit for a maximally pressure-staged LPA is three stages for a single plan view size. Figure 7 shows the gain characteristics for a three-stage gain block with $\sigma_1=3.6$, $\sigma_2=1.2$, and $\sigma_3=0.4$ with $b_g=0.5$ mm. (Appendix A has printouts of nominal predicted and design operation, and appendix B lists stacking orders.)

This high gain unit has an excellent null characteristic, linearity, and gain. However, the first stage operates at a very low supply pressure ($P_g = 0.3$ torr) and consequently limits the bandpass to approximately 117 Hz. In addition, the output resistance's being considerably higher than the input (40 to 1) makes it difficult to use this device in a feedback configuration.

By compromising in the staging makeup, other configurations are possible. The three following gain blocks illustrate such compromises.

High-input-impedance gain block.—In an effort to increase the input impedance and the bandpass of the gain block, the first—and second—stage aspect ratios can be reduced. Figure 8 shows data taken for a gain block made up of $\sigma_1 = 0.75$, $\sigma_2 = 0.5$, $\sigma_3 = 0.4$, and $\sigma_3 = 0.5$ mm. This typical gain block is used as a preamplifier for laminar jet angular rate sensors and sometimes for the photofluidic interface. (The photofluidic interface most commonly uses a second—stage aspect ratio, $\sigma = 0.6$, but this use does not materially change the characteristics from the block discussed here.)

Gain is about half of the maximum dynamic range device, but the output-to-input resistance ratio is reduced to about 3.

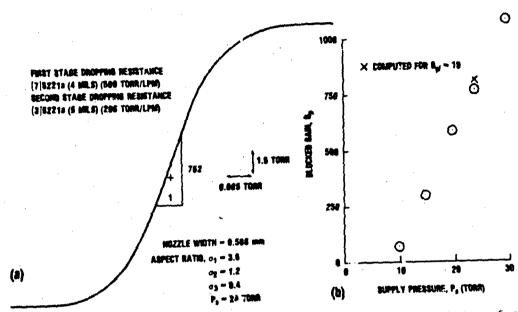


Figure 7. Three-stage maximum dynamic range gain block: (a) transfer and (b) gain characteristics.

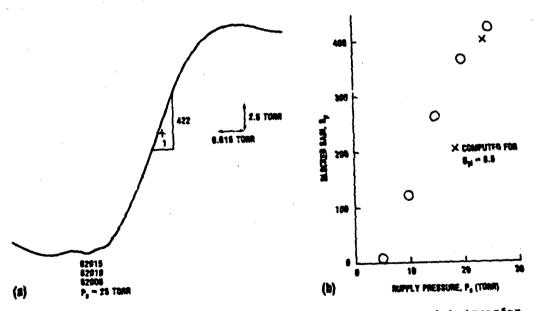


Figure 8. Three-stage high-input-impedance gain block: (a) transfer and (b) gain characteristics.

Change-in-size moderate-pressure staging gain block.--In a modification of the common supply pressure gain block with varying sizes, the gain can be increased by lowering the aspect ratio of the final stage. This device requires only one dropping resistor to supply the common first- and second-stage pressure. (The resistor must now accommodate flow for two stages and must be appropriately sized so that the flow per laminate does not exceed 0.1 LPM.) Shown in figure 9 are the characteristics for a gain block mode up of 63020, 62020, and 81008 LPA's.

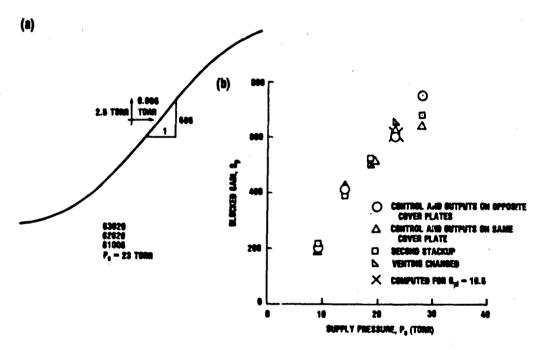


Figure 9. Three-stage preamplifier for laminar jet angular rate sensor: (a) transfer and (b) gain characteristics.

The dropping resistor is required to drop 24 to 4 torr or 20 torr at a flow rate of 0.75 LPM or R=27 torr/LPM. This drop is readily achieved with eight 200-torr/LPM resistors.

The gain of 605 compares favorably with the design goal of 608 from the program in appendix A at G=10.5 and is 20 to 25 percent higher than the previous case. However, the input-to-output resistance ratio is worse and the bandwidth is down.

Flueric servovalve amplifier.—The flueric servovalve 11,12 is a three-stage LPA with multiple parallel elements in each stage. As configured in C format, it consists of a first stage containing two 63020 parallel elements, a second stage containing three 62010 parallel elements, and a third stage containing six 61505 parallel elements. The design program predicts a gain of 335, a bandpass of 281 Hz, a leak flow of the first two stages of 1.757 LPM, and an output-to-input resistance ratio of 9.5 (example A-8). Figure 10 shows a Bode plot taken at the Massachusetts Institute of Technology for operation in hydraulic oil. (Note that the volumetric flows in air and oil are the same, the pressures are different, and the dynamics are esentially the same since the kinematic viscosities are similar.) Agreement between the prediction of the design program and the data is good.

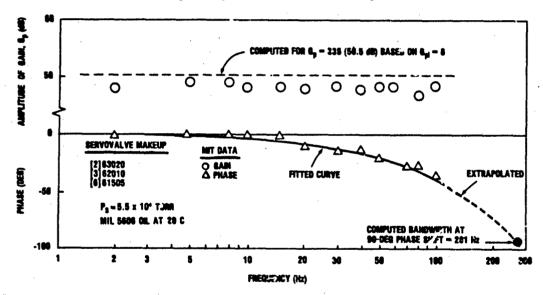


Figure 10. Bode plot of flueric servovalve three-stage gain block.

Five parameters are of interest in the design of a flueric servovalve:

- leakage flow
- input-to-output
- bandpass
- resistance ratio
- gain
- pressure recovery

¹¹ D. N. Wormley, D. Lee, and K. M. Lee, Development of a Fluidic, Hydraulic Servovalve, Massachusetts Institute of Technology, contract with Harry Diamond Laboratories, HDL-CR-81-216-1 (February 1981).

¹²D. Lee and D. N. Wormley, A Fluidic Rydraulic Servovalve, J. Dyn. Sys. Neas. Control, 103, No. 4 (December 1981).

Given the plan view of the standard LPA and the sizes available, the thinnest lamination with a reasonable aspect ratio gives rise to either $\sigma=0.375$ for $b_g=0.375$ mm or $\sigma=0.5$ for $b_g=0.25$ mm so that operation in air at 64 torr is possible, corresponding to 5.5×10^4 torr in oil (1100 psi). Pressure recovery is thus fixed. Pressure recovery can be increased by changing LPA geometry, but that increase is not within the purvious of this report. Thus, let us examine the optimization or increase in performance of the other three parameters at the expense, if any, of gain. By using the design program, some changes can be easily tried out. Table 1 shows design parameters as they are affected by various changes.

TABLE 1. OPTIMIZATION OF FLUERIC SERVUVALVE FOR LEAKAGE FLOW, BANDPASS, OUTPUT-TO-INPUT RESISTANCE RATIO, AND GAIN

Trial		block keup	Geometric changes from previous case	Leakage flow (LPM)	Bandpass (Hz)	Output-to-input resistance ratio	Gain
1	[3]63020]62010]61503	(Baseline)	1.76	281	9.5	335
2	[4] 62020] 62010] 51005	$ \sigma_1 = 1 $ $ b_{a1} = 0.5 \text{ mg} $ $ N_1 = 3 $ $ N_2 = 4 $ $ N_3 = 10 $ $ b_{a3} = 0.254 \text{ mg} $ $ \sigma_3 = 0.5 $	2.07	376	7.2	320
3	(3)	62020 62010 61505	$N_1 = 2$ $N_2 = 3$ $D_{03} = 0.37 \text{ mm}$	1.49	363	4.0	285
4	[2]	62016 62010 61505	σ ₁ = 0.8 N ₂ = 2 N ₃ = 12	1.19	412	2.1	229
5	[3]	62016 62010 61505	N ₁ = 3 N ₂ = 3 N ₃ = 8	1.79	412	4.7	289
6	[3]	62016 62010 61505	N, = 4 N ₃ = 6	2.09	412	9.5	321
7	(2)	62020 62010 61505	N ₁ = 2 N ₂ = 2 \sigma_1 = 1 N ₃ = 8	1.19	363	5.0	289

Notes: σ = aspect ratio, b_{x} = supply nozzle width, N = number of stages.

In trying to follow the rationale for the changes shown in the second column of table 1, the following arguments are set forth.

• First change (trial 2).--Increase the first-stage bandwidth by decreasing LPA size, but keep the input resistance the same by using more elements. Decrease the output resistance by increasing the number of output elements to 10. Keep the gain up by reducing second-stage resistance relative to the third-stage resistance by increasing the number of parallel elements in the second stage and decreasing element size in the third stage.

Result.--The bandpass increases at the expense essentially of leakage flow.

• Second change (trial 3).--Increase the input resistance by reducing the number of elements in the first stage. Decrease the output resistance by increasing the element size back up to baseline. Reduce leakage by reducing the number of elements in the second stage.

Result.--Leakage flow and the output-to-input resistance ratio decrease at the expense of gain.

• Third change (trial 4).--Further increase the input resistance by reducing the first stage aspect ratio to 0.8, thereby also increasing the bandpass. Further decrease leakage flow by reducing the number of elements in the second stage to two. Decrease the output resistance by increasing the number of output elements to 12.

Result.--Leakage flow and the output-to-input resistance ratio decrease and the bandpass increases at the further expense of gain.*

• Fourth change (trial 5).—Bring the gain back up by reducing the input resistance by adding an element to the first stage, increase the output resistance by decreasing the number of output elements, and increase the second- to third-stage gain by increasing the number of second-stage elements to three.

³T. M. Drzewiecki, D. Wormley, and F. M. Manion, A Computer-Aided Pesign Procedure for Laminar Fluidic Systems, J. Dyn. Sys. Meas. Control, 97, Series G, No. 4 (December 1975).

^{*}Reduction in gain is not altogether unacceptable when it is noted that in a feedback arrangement the overall performance is dependent somewhat on closed loop gain, which increases with a decreasing output-to-input resistance ratio. Since gain is down 32 percent but the input-to-output resistance ratio is up over 400 percent, this case may be more desirable.

Result.--Leakage flow is back up to baseline, but the bandpass is still high at a slight expense to the input-to-output resistance ratio. The gain is up, but is still down from the baseline.

• fifth change (trial 6).--Examine the condition in which the output-to-input resistance ratio bandpass is high, by increasing the number of elements in the first stage to fur and in the third stage to six.

Result.--The bandpass is high at the expense of leakage flow. The gain is back at the baseline.

• Final change (trial 7).--Decrease leakage flow by reducing the number of elements in the first and second stages to two each. Keep the cain up by increasing the tirst-stage aspect ratio to one. Decrease the output resistance by increasing the number of output elements to eight.

Result.--Leakage is reduced to 68 percent of the baseline, the bandpass is increased by 29 percent over the baseline, and the output-to-input resistance ratio is halved, with only a 14-percent penalty in gain.

preliminary tests at the Massachusetts Institute of Technology have verified this capability to increase bandwidth and decrease leakage flow.

3.2.4 Power- or Flow-Gain Gain Blocks

In some applications, pressure gain is not important. However, the ability to deliver flow at a given pressure is important when, for example, a positive displacement actuator must be moved at a given rate. It is clear from the arguments presented above that output impedance can be significantly reduced by employing mulciple parallel elements. One gain block that has been tried for driving small actuators involves a common supply pressure arrangement consisting of LPA's with $b_{\rm S}=0.5$ mm and $\sigma=0.4$ with a first stage of three parallel elements, a second stage of four, and a third stage of eight. The unit makeup is designated as [3]62008, [4]62008, [8]62008. The measured gain of 75 compares favorably with the $^{\circ}0$ computed by using a single change gain of 8.

The output resistance of 5.84 torr/LPM indicates that at the roughly 6-torr maximum power point the unit can deliver about 1 LPM of flow.

3.3 Four-Stage Gain Blocks

Gain blocks with more than three stages are not very common in dc applications primarily because as the gain becomes high, even a small initial null offset can be devastating. In ac applications where highpass filters are commonly used, a dc null offset does not affect operation. One such example is in the fluidic intercom system described by this author. 5 In that case, a four-stage LPA is composed of a three-stage, self-staged preamplifier of $b_{\rm g}=0.25$ mm and $\sigma=1.2$ and a power fourth stage composed of two parallel elements with $b_{\rm g}=0.25$ mm and $\sigma=0.8$. The gain of this device is about 60 dB for 1 to 2 kHz.

One dc application in which a four-stage device is used, however, is a pressure regulator.* Since it is used in a feedback configuration, it is important that the input and output resistances be comparable, that the gain be relatively high, and that it be capable of delivering sufficient flow to a load at, in this case, 6 torr. For these reasons, at least four stages are required. The gain block designed for this application is composed of a first stage of one element of $b_{s1} = 0.5$ mm and $\sigma = 0.75$ (62015), a second stage of two elements of $b_{\rm S,2} = 0.375$ mm and $\sigma_{\rm 2} = 0.67$ ([2]61510), a third stage of three elements of $b_{s3} = 0.375$ mm and $a_{s3} = 0.67$ ([3]61510), and a fourth stage of six elements of $b_{s4} = 0.375$ mm and $\sigma_4 = 0.33$ ([6]61505). The transfer characteristics and the gain as a function of supply pressure to the last stage are shown in figure 11 (p. 26). The very low gain of 61505 LPA's is the reason that the average single-stage gain is 7.5. The gain would be considerably increased by using components with better gain, say, 9 to 10. This use would increase the gain 100 to 200 percent.

3.4 Five-Stage Gain Block for Public Address System

The only currently used five-stage gain block is in a flueric public address system at HDL. This device is designed to have high gain and bandpass and very low output impedance in order to drive large exponential horns. To provide good matching between stages, it was decided to operate each stage at the same impedance despite changes in size or supply pressure. The design program was exercised with the result shown in figure 12. As can be seen, three different size LPA's

⁵T. M. Drzewiecki, A Fluidic Audio Intercom, Proceedings of 20th Anniversary of Fluidics Symposium, American Society of Mechanical Engineers special publication G00177, Chicago, IL (November 1980).

^{*}T. M. Drzewiecki, Some Active Fluidic Compensation Circuits, submitted for American Society of Mechanical Engineers, 1982 Winter Annual Meeting, Phoenix, AZ (November 1982).

are used in an almost perfect impedance matching scheme. The dc gain was not available since the device is high passed between the first and second stages and between the fourth and fifth stages with inductive shunts designed for passing a frequency of 600 Hz. The device works well and with minimum distortion over a 600- to 3000-Hz frequency band.

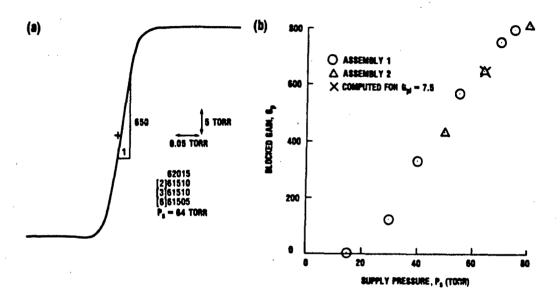


Figure 11. Four-stage pressure regulator gain block: (a) transfer and (b) gain characteristics.

3.5 Eight-Stage Gain Block

Andrew Commencer of the Commencer of the

There are no commonly used gain blocks of six or seven stages. However, there is one device of eight stages that was used in the stabilization system of an M48 tank. 13 This device was designed by Charles Paras (formerly of HDL) and as implemented used some turbulent final stages. Table 2 shows the makeup.

The design program does not allow σN_R to be any other value than 1000. However, since the individual gains are not greatly different from their nominal values and only the resistances change in the final two stages, it should not give a bad estimate. For an individual

¹³C. L. Abbott, T. B. Tippetts, S. M. Tenney, and C. Paras, A Study of Fluidic Gun Stabilization Systems for Combat Vehicles: Final Report, Airesearch Manufacturing Company of Arizona, contract with Harry Diamond Laboratories, HDL-CR-80-100-1 (April 1980).

```
IS YOUR LPA GAIN 10 ? YES
NUMBER OF STAGES? 5
          ASPECT RATIO? .75
STAGE
NO. OF PARALLEL ELEMENTS IN STAGE 1 2 1
STAGE
           NOZZLE WIDTH IN MM? .508
STACE
           SUPPLY PRESSURE = 6.888902666694
                                               TORR
STAGE
           SUPPLY FLOW = .3110204081646 LPM
STAGE
           SUPPLY RESISTANCE = 22.14935896762 TORR/LPM
STAGE 2 ASPECT RATIO? .5
NO. OF PARALLEL ELEMENTS IN STAGE 2 ? 2
           NOZZLE WIDTH IN MMP .508
STACE
STAGE
           SUPPLY PRESSURE = 15.50003100006
           SUPPLY FLOW = .5644444444422 LPM
STAGE
STAGE
           SUPPLY RESISTANCE = 27.46068484274 TORR/LPM
STAGE
           ASPECT RATIO? .66667
NO. OF PARALLEL ELEMENTS IN STAGE 3 ? 3
STAGE
           NOZZLE WIDTH IN MMP .38t
          SUPPLY PRESSURE = 15.49987600082
STAGE
STAGE
           SUPPLY FLOW = .6858006858039 LPM
STAGE
           SUPPLY RESISTANCE = 22.60113808818 TORR/LPM
           ASPECT RATIO? 1
STAGE
NO. OF PARALLEL ELEMENTS IN STAGE: 4 ? 4
STAGE
          NOZZLE WIDTH IN MM? .. 254
STAGE
          SUPPLY PRESSURE = 15.50003100006 TORR
          SUPPLY FLOW = .6349999999976 LPM
STAGE
STAGE
          SUPPLY RESISTANCE = 24.40949763798 TORR/LPM
          ASPECT RATIO? .5
STAGE
NO. OF PARALLEL ELEMENTS IN STAGE 5 ? 20
STAGE
          NOZZLE WIDTH IN MM? "254
STAGE
          SUPPLY PRESSURE = 62.00012400025
          SUPPLY FLOW = 2.822022222212 LPM
STAGE
STAGE
          SUPPLY RESISTANCE = 21.96854787418 TORR/LPM
FIRST STAGE BIAS PRESSURE IN TORR? O
ENTER DUTPUT LOAD IN TERMS OF AN EQUIVALENT LPA
A BLOCKED LOAD IS PUT IN AS AN AR=1 AND BS=0
LOAD NO. OF PARALLEL STAGES? 1
LOAD ASPECT RATIO? 1
LOAD NOZZLE WIDTH(MM)? O
STAGE
          PRESSURE GAIN = 6.50198537992
       1
STAGE
          PRESSURE GAIN ==
                            5.072692711794
          PRESSURE GAIN =
STAGE
       7
                            5.151701658716
STAGE
          PRESSURE GAIN =
                            4.786223675775
       5 PRESSURE GAIN = 9.523809523819
PRESSURE GAIN OF THE 5 STAGE GAINBLOCK = 7745.312407647
GAINBLOCK INPUT RESISTANCE = 16.61201922572 TORR/LPM GAINBLOCK OUTPUT RESISTANCE = 10.98427393709 TORR/LPM
GAINBLOCK BANDWIDTH AT 90DEG PHASE SHIFT = 305.1565742289 HERTZ
DO YOU WANT TO MAKE ANOTHER CALCULATION? NO
```

Figures 12. Design program output for Harry Diamond Laboratories flueric public address system.

nominal gain of 10 per stage, the program computes a value of gain of 2.72 \times 10⁶ (example A-9). This is in excellent agreement with the measured value of 3 \times 10⁶.

This then indicates the power of the design program for estimeting off-design points.

TABLE 2. MAKEUP OF EIGHT-STAGE GAIN BLOCK FOR M48A5 GUN STABILIZATION SYSTEM

Stage	Aspect ratio	Nozzle width (mm)	oNR
1	1.07	0.76	701
2	1.00	0.76	983
3	0.83	0.76	884
4	0.67	0.76	977
5 6 7 8	0.53 0.33 0.45 0.30	0.76 0.76 0.51 0.76	957 1403 4000 4617

Note: oN_R = aspect ratio times Reynolds number.

4. DISCUSSION AND CONCLUSIONS

This report presents gain and transfer characteristic data for most of the currently used gain blocks. These gain blocks have all been implemented in C format using standard HDL LPA's.

During collection of data, it was discovered that certain LPA's did not conform to the desired characteristics of bias insensitivity as designed 7 in the original HDL model 3.1.1.8. These were specifically the fineblanked devices. Although they operated well and pretty much as predicted in pressure-staged configurations, they operated poorly in self-staged configurations. This behavior extended to metal etched parts manufactured from the same program that made the fineblanking dies. Subsequently, LPA's designed specifically to HDL dimensions were made by electrodischarge machining and metal etching and were found to be as desired. Consequently, no fineblanked parts exist, at this writing, that are suitable for proper high-bias or self-staged operation.

The computer design-guide program has been shown to be an effective tool for gain-block design, optimization, and off-design estimation. Optimization has been specifically demonstrated on the flueric servo-valve amplifier.

⁷T. M. Drzewiecki, Fluerics 38: A Computer-Aided Design Analysis for the Static and Dynamic Port Characteristics of Laminar Proportional Amplifiers, Harry Diamond Laboratories, HDL-TR-1758 (June 1976).

Interestingly, high gain can be implemented with considerably fewer stages if one uses positive feedback. Typically, gains of 150 to 500 can be obtained in two stages at the expense of bandwidth. By using the first amplifier as a buffer or preamplifier and a second stage with feedback, the loss in bandwidth can be minimized. An example of using positive feedback is the three-stage servovalve in which gain is increased from 335 to over 1500.

Finally, design guidelines for amplifier stacking and implementation and a format for stacking orders have been presented.

The reader is encouraged to use the results presented here as a basis for a private library of gain blocks. Other useful devices should be logged, documented, and reported, as soon as practicable.

ACKNOWLEDGEMENTS

This report would have been far leaner for data had it not been for the diligence and hard work of our co-op student, Bill O'Brien, who was responsible for generating most of the gain information. The able assistance and encouragement of James Joyce also are greatly appreciated.

LITERATURE CITED

- (1) F. M. Manion and G. Mon, Fluerics 33: Design and Staging of Laminar Proportional Amplifiers, Harry Diamond Laboratories, HDL-TR-1608 (September 1972).
- (2) F. M. Manion and T. M. Drzewiecki, Analytical Design of Laminar Proportional Amplifiers, Proceedings of HDL Fluidic State-of-the-Art Symposium, 1 (October 1974).
- (3) T. M. Drzewiecki, D. Wormley, and F. M. Manion, A Computer-Aided Design Procedure for Laminar Fluidic Systems, J. Dyn. Sys. Meas. Control, 97, Series G, No. 4 (December 1975).
- (4) G. Mon, Flueric Laminar Gain Blocks and an Operational Amplifier Scaler, Harry Diamond Laboratories, HDL-TR-1730 (December 1975).
- (5) T. M. Drzewiecki, A Fluidic Audio Intercom, Proceedings of 20th Anniversary of Fluidics Symposium, American Society of Mechanical Engineers special publication G00177, Chicago, IL (November 1980).
- (6) M. Cycon and D. Shaffer, Design Guide for Laminar Flow Pluidic Amplifiers and Sensors, contract with Garrett Pneumatic Systems Division, Harry Diamond Laboratories, HDL-CR-82-288-1 (Narch 1982).
- (7) T. M. Drzewiecki, Fluerics 38: A Computer-Aided Design Analysis for the Static and Dynamic Port Characteristics of Laminar Proportional Amplifiers, Harry Diamond Laboratories, HDL-TR-1758 (June 1976).
- (8) L. Scheer, J. Roundy, and J. Joyce, Manufacturing Techniques for Producing High Quality Fluidic Laminates in Production Quantities, Proceedings of 20th Anniversary of Fluidics Symposium, American Society of Mechanical Engineers special publication G00177, Chicago, IL (November 1980).
- (9) R. M. Phillippi, A Study of Fineblanking for the Manufacture of Flueric Laminar Proportional Amplifiers, Harry Diamond Laboratories, HDL-TM-77-8 (May 1977).
- (10) R. M. Phillippi, T. M. Drzewiecki, T. Negas, and H. S. Parker, Design of a Fluidic Capillary Pyrometer for Contact Duty at Temperatures to 2750°C, Proceedings of 6th International Symposium on Temperature, Washington, DC (15 to 19 March 1982).

- (11) D. N. Wormley, D. Lee, and K. M. Lee, Development of a Fluidic, Hydraulic Servovalve, Massachusetts Institute of Technology, contract with Harry Diamond Laboratories, HDL-CR-81-216-1 (February 1981).
- (12) D. Lee and D. N. Wormley, A Fluidic Hydraulic Servovalve, J. Dyn. Sys. Meas. Control, 103, No. 4 (December 1981).
- (13) C. L. Abbott, T. B. Tippetts, S. M. Tenney, and C. Paras, A Study of Fluidic Gun Stabilization Systems for Combat Vehicles: Final Report, Airesearch Manufacturing Company of Arizona, contract with Harry Diamond Laboratories, HDL-CR-80-100-1 (April 1980).

APPENDIX A.--BASIC PROGRAM AND REPRESENTATIVE EXAMPLES

This appendix contains the computer program, written in BASIC, that yields design approximations for multistage laminar proportional amplifiers. The program is based on a single-stage block-loaded gain of 10, but allows the user to insert different corresponding gain values where appropriate.

Following a listing of the program, nine different sample runs are presented.

```
10 DIM P(10).Q(10),K(10),A(10),B(10),G(10)
20 G0=10
 30 PRINT
 40 PRINT "THIS PROGRAM GIVES DESIGN APPROXIMATIONS FOR MULTISTAG
E LPAS"
50 PRINT "BASED ON THE STANDARD HOL LPA. ASSUMING A ZERO-BIAS.BL
OCKED"
40 PRINT "SINGLE STAGE GAIN OF 10 OPERATING AT SIGHA*NR±1000 IN
70 PRINT "FOR MAXIMUM DYNAMIC RANGE THE PRODUCT OF ASPECT RATIO AND NOZZLE WIDTH OF EACH SUCCEDING STAGT SHOULD BE 1/3RD OF THE PREVIOUS, E.G.FOR BS1=BS2.AR(1)=3 AND AR(2)=1."
90 PRINT
100 INFUT "IS YOUR LPA GAIN 10 ", YS: IF STR(YS.1,1) = "Y"THEN 120
110 INPUT "WHAT IS YOUR GAIN", GO
120 INPUT "NUMBER OF STAGES".N
130 FOR M=1TO N
130 FRINT "STAGE ";M;" ASPECT RATIO";:INPUT A(M+1)
150 FRINT "NO. OF PARALLEL ELEMENTS IN STAGE ";M;:INPUT K(M+1)
160 FRINT "STAGE ";M;" NOZZLE WIDTH IN MM";:INPUT B(M+1)
170 P(H+1)=1/(A(H+1)#B(H+1))+2
180 PRINT "STAGE ":M; " SUPPLY PRESSURE = ";P(M+1); " TORR"
190 Q(M+1)=2.5#SQR(P(M+1))#B(M+1)+2/(1+1/A(M+1))+2*K(M+1)
200 PRINT "STAGE ":M:" SUPPLY FLOW =":Q(M+1);"LPM"
210 PRINT "STAGE ":M;" SUPPLY RESISTANCE =";P(M+1)/Q(M+1);"TORR/
LPM'
220 NEXT H
230 INPUT "FIRST STAGE BIAS PRESSURE IN TORR". PO
240 P(1)=P0#3
250 PRINT "ENTER OUTPUT LOAD IN TERMS OF AN EQUIVALENT LPA"
260 PRINT "A BLOCKED LOAD IS PUT IN AC AN AR=1 AND BS=0"
270 INPUT "LOAD NO. OF PARALLEL STAGES", K(N+2)
280 INPUT "LOAD ASPECT RATIO", A(N+2)
290 INPUT "LOAD NOZZLE WIDTH(MM)", B(N+2)
300 IF B(N+2)=0 THEN 310:P(N+2)=1/(A(N+2)*B(N+2))+2:Q(N+2)=2.5*S
QR(P(N+2))*K(N+2)*B(N+2)+2/(1+1/A(N+2))+2:GOTO 320
310 P(N+2)=1:Q(N+2)=0
320 L=0
330 FOR J=1 TO N
340 G(J)=G04(1/(1+.20#P(J)/P(J+1)))#(1/(1+.667#P(J+1)#G(J+2)/P(J
+2)/0(J+1)))
350 PRINT "STAGE ": J: " PRESSURE GAIN = ":G(J)
360 L=L+B(J+1)/SQR(F(J+1))
370 IF J=1THEN 400
380 G=G(J)#G
390 GOTO 410
400 G=G(J)
410 NEXT J
420 PRINT "PRESSURE GAIN OF THE ";N;" STAGE GAINBLOCK =";G
430 PRINT "GAINBLOCK INPUT RESISTANCE = ";.75#P(2)/G(2):"TORR/LP
440 PRINT "GAINGLOCK OUTPUT RESISTANCE = ";.5#P(N+1)/Q(N+1);"TOR
R/LPM
450 PRINT "GAINBLOCK BANDWIDTH AT 90DEG PHASE SHIFT = ";157.5/L;
"HERTZ"
460 INPUT "DO YOU WANT TO MAKE ANOTHER CALCULATION", 18: IF $TR(18
,1,1)="Y" THEN 470:GOTO 480
470 PRINT :PRINT :PRINT :GOTO 120
480 STOP : END
```

BASIC PROGRAM

THIS PROGRAM GIVES DESIGN APPROXIMATIONS FOR MULTISTAGE LPAS
BASED ON THE STANDARD HDL LPA. ASSUMING A ZERO-BIAS.BLOCKED

BASED ON THE STANDARD HOL LPA, ASSUMING A ZERO-BIAS, BLOCKED SINGLE STAGE GAIN OF 10 OPERATING AT SIGMA*NR=1000 IN AIR.

FOR MAXIMUM DYNAMIC RANGE THE PRODUCT OF ASPECT RATIO AND NOZZLE WIDTH OF EACH SUCCEDING STAGE SHOULD BE 1/3RD OF THE PREVIOUS, E.G.FOR BS1=BS2, AR(1)=3 AND AR(2)=1.

IS YOUR LPA GAIN 10 ? N WHAT IS YOUR GAIN? 8 NUMBER OF STAGES? 2 ASPECT RATIO? .667 NO. OF PARALLEL ELEMENTS IN STAGE 1 ? 1 STAGE 1 NOZZLE WIDTH IN MM? .381 SUPPLY PRESSURE = 15.4845425861 STAGE SUPPLY FLOW = .2286228417162 LPM STAGE SUPPLY RESISTANCE = 67.72963921655 TORR/LPM STAGE ASPECT RATIO? .667 STAGE NO. OF PARALLEL ELEMENTS IN STAGE 2 ? 1 STAGE NOZZLE WIDTH IN MM? .381 **SUPPLY PRESSURE = 15.4845425861** STAGE STAGE SUPPLY FLOW = .2286228417162 LPM SUPPLY RESISTANCE = 67.72963921655 TORR/LPM FIRST STAGE BIAS PRESSURE IN TORR? O ENTER OUTPUT LOAD IN TERMS OF AN EQUIVALENT LPA A BLOCKED LOAD IS PUT IN AS AN AR=1 AND BS=0 LOAD NO. OF PARALLEL STAGES? 1 LOAD ASPECT RATIO? 1 LOAD NOZZLE WIDTH(HM)? 0 PRESSURE GAIN = 4.799040191962 PRESSURE GAIN = 6.6666666666 STAGE PRESSURE GAIN OF THE 2 STAGE GAINBLOCK = 31.99360127974 GAINBLOCK INPUT RESISTANCE = 50.79722941243 TORR/LPM GAINBLOCK OUTPUT RESISTANCE = 33.86481960827 TORR/LPM GAINBLOCK BANDWIDTH AT 90DEG PHASE SHIFT = 813.3449550178 HERTZ DO YOU WANT TO MAKE ANOTHER CALCULATION? N

STOP

Example A-1. Two-stage, self-staged gain block.

THIS PROGRAM GIVES DESIGN APPROXIMATIONS FOR MULTISTAGE LPAS BASED ON THE STANDARD HDL LPA, ASSUMING A ZERO-BIAS, BLOCKED SINGLE STAGE GAIN OF 10 OPERATING AT SIGMA*NR=1000 IN AIR.

FOR MAXIMUM DYNAMIC RANGE THE PRODUCT OF ASPECT RATIO AND NOZZLE WIDTH OF EACH SUCEEDING STAGE SHOULD BE 1/3RD OF THE PREVIOUS, E.G.FOR BS1=BS2, AR(1)=3 AND AR(2)=1.

```
IS YOUR LPA GAIN 10 ? N WHAT IS YOUR GAIN? 8.5
NUMBER OF STAGES? 2
           ASPECT RATIO? .66667
STAGE 1
NO. OF PARALLEL ELEMENTS IN STAGE 1 ? 1
          NOZZLE WIDTH IN MM? .381
SUPPLY PRESSURE = 15.49987600082
STAGE
       1
STAGE
           SUPPLY FLOW = .2286002286013 LPM
STAGE
           SUPPLY RESISTANCE = 67.80341426453 TORR/LPM
STAGE
STAGE
           ASPECT RATIO? .333333
NO. OF PARALLEL ELEMENTS IN STAGE
STAGE
           NOZZLE WIDTH IN MM? .381
           SUPPLY PRESSURE = 62.00024800074
STAGE
           SUPPLY FLOW = .1785936607044 LPM
STAGE
          SUPPLY RESISTANCE = 347.1581676315 TORR/LPM
STACE
FIRST STAGE BIAS PRESSURE IN TORR? O
ENTER OUTPUT LOAD IN TERMS OF AN EQUIVALENT LPA
A BLOCKED LOAD IS PUT IN AS AN AR=1 AND BS=0
LOAD NO. OF PARALLEL STAGES? 1
LOAD ASPECT RATIO? 1
LOAD NOZZLE WIDTH(MM)? O
       1 PRESSURE GAIN = 7.520315831375
2 PRESSURE GAIN = 8.09524272106
STAGE
PRESSURE GAIN OF THE 2 STAGE GAINBLOCK = 60.87878199401
GAINBLOCK INPUT RESISTANCE = 50.85256069842 TORR/LPM
GAINBLOCK OUTPUT RESISTANCE = 173.5790838157 TORR/LPM
GAINBLOCK BANDWIDTH AT 90DEG PHASE SHIFT = 1084.998915014 HERTZ
DO YOU WANT TO MAKE ANOTHER CALCULATION? N
```

Example A-2. Two-stage gain block of fluidic capillary pyrometer.

STOP

THIS PROGRAM GIVES DESIGN APPROXIMATIONS FOR MULTISTAGE LPAS BASED ON THE STANDARD HOL LPA, ASSUMING A ZERO-BIAS, BLOCKED SINGLE STAGE GAIN OF 10 OPERATING AT SIGMA*NR=1000 IN AIR.

FOR MAXIMUM DYNAMIC RANGE THE PRODUCT OF ASPECT RATIO AND NOZZLE WIDTH OF EACH SUCCEDING STAGE SHOULD BE 1/3RD OF THE PREVIOUS. E.G.FOR BS1=B°2, AR(1)=3 AND AR(2)=1.

IS YOUR LPA GAIN 10 ? Y NUMBER OF STAGES? 3 ASPECT RATIO? 1 STAGE NO. OF PARALLEL ELEMENTS IN STAGE NOZZLE WIDTH IN MM? .508 STAGE 1 BTAGE SUPPLY PRESSURE = 3.875007750016 TORR STAGE SUPPLY FLOW = .3174999999988 LPM SUPPLY RESISTANCE = 12.20474881899 TORR/LPM STAGE ASPECT RATIO? 1 STAGE NO. OF PARALLEL ELEMENTS IN STAGE 2 ? 1 NOZZLE WIDTH IN MM? .508 STAGE SUPPLY PRESSURE = 3.875007750016 STAGE SUPPLY FLOW = .3174999999988 LPM STAGE STAGE SUPPLY RESISTANCE = 12.20474881899 TORR/LPM ASPECT RATIO? 1 STAGE NO. OF PARALLEL ELEMENTS IN STAGE NOZZLE WIDTH IN MM? .508 STAGE SUPPLY PRESSURE = 3.875007750016 STAGE STAGE SUPPLY FLOW = .3174999999988 LPM SUPPLY RESISTANCE = 12.20474881899 TORR/LPM STACE FIRST STAGE BIAS PRESSURE IN TORR? O ENTER OUTPUT LOAD IN TERMS OF AN EQUIVALENT LPA A BLOCKED LOAD IS PUT IN AS AN AR=1 AND BS=0 LOAD NO. OF PARALLEL STAGES? 1 LOAD ASPECT RATIO? 1 LOAD NOZZLE WIDTH(MM)? O STAGE PRESSURE GAIN = 5.998800239952 PRESSURE GAIN = 4.99900019996 STAGE PRESSURE GAIN = 8.333333333333 PRESSURE GAIN OF THE 3 STAGE GAINBLOCK = 249.900029992 GAINBLOCK INPUT RESISTANCE = 9.153561614246 TORR/LPM GAINBLOCK OUTPUT RESISTANCE = 6.102374409497 TORR/LPM GAINBLOCK BANDWIDTH AT 90DEG PHASE SHIFT = 203.437906875 HERTZ DO YOU WANT TO MAKE ANOTHER CALCULATION? Y

Example A-3. Three-stage, self-staged gain block.

```
NUMBER OF STAGES? 3
STAGE 1 ASPECT RATIO? .6667
NO. OF PARALLEL ELEMENTS IN STAGE
STAGE
           NOZZLE WIDTH IN MM? .76
       1
           SUPPLY PRESSURE = 3.895039849219
SUPPLY FLOW = .4560045596449 LPM
STAGE
STAGE
           SUPPLY RESISTANCE = 8.54166864527 TORR/LPM
STAGE
STAGE
           ASPECT RATIO? 1
NO. OF FARALLEL ELEMENTS IN STAGE
STAGE
           NOZZLE WIDTH IN MM? .508
           SUPPLY PRESSURE = 3.875007750016
STAGE
STAGE
           SUPPLY FLOW = .3174999999988 LPM
           SUPPLY RESISTANCE = 12.20474881899 TORR/LPM
STAGE 2
           ASPECT RATIO? 2
STAGE
NO. OF PARALLEL ELEMENTS IN STAGE 3 ? 1
           NOZZLE WICTH IN MM? .254
STAGE
       3
STAGE
           SUPPLY PRESSURE = 3.875007750016
STAGE
           SUPPLY FLOW = .1411111111105 LPM
STAGE
          SUPPLY RESISTANCE = 27.46068484275 TORR/LPM
      3
FIRST STAGE BIAS PRESSURE IN TORR? O
ENTER OUTPUT LOAD IN TERMS OF AN EQUIVALENT LPA
A BLOCKED LOAD IS PUT IN AS AN AR=1 AND BS=0
LOAD NO. OF PARALLEL STAGES? 1
LOAD ASPECT RATIO? 1
LOAD NOZZLE WIDTH(MM)? O
          PRESSURE GAIN = 6.817517773483
PRESSURE GAIN = 6.422303400955
STAGE
STAGE
      3 PRESSURE GAIN = 8.333333333333
STAGE
PRESSURE GAIN OF THE 3 STAGE GAINBLOCK = 364.8680631895
GAINBLOCK INPUT RESISTANCE = 6.406251483952 TORR/LPM
GAINBLOCK OUTPUT RESISTANCE = 13.73034242139 TORR/LPM
GAINBLOCK BANDWIDTH AT 90DEG PHASE SHIFT = 203.9674795827 HERTZ
DO YOU WANT TO MAKE ANOTHER CALCULATION? Y
```

Example A-4. Three-stage, common supply gain block.

```
SINGLE STAGE GAIN OF 10 OPERATING AT SIGMA*NR=1000 TH AIR.
FOR MAXIMUM DYNAMIC RANGE THE PRODUCT OF ASPECT RATIO AND
NOZZLE WIDTH OF EACH SUCCEDING STAGE SHOULD BE 1/3RD OF THE
PREVIOUS, E.G.FOR BS1=BS2, AR(1)=3 AND AR(2)=1.
IS YOUR LPA GAIN 10 ? Y
NUMBER OF STAGES? 3
STAGE
          ASPECT RATIO? 1
NO. OF PARALLEL ELEMENTS IN STAGE
          NOZZILE WYDTH IN MM? .254
STAGE
          SUPPLY PRESSURE = 15.50003100006
STAGE
          SUPPLY FLOW = .634799999976 LPM
STAGE
STAGE
          SUPPLY RESISTANCE = 24.40949763798 TORR/LPM
          ASPECT RATIO? 1
STAGE
NO. OF PARALLEL ELEMENTS IN STAGE 2 ? 2
          NOZZLE WIDTH IN MM? .254
STAGE
          SUPPLY PRESSURE = 15.50003100006
STAGE
          SUPPLY FLOW = .3174999999988 LPM
STAGE
STAGE
          SUPPLY RESISTANCE = 48.81899527596 TORR/LPM
STAGE
       3
          ASPECT RATIO? 1
NO. OF PARALLEL ELEMENTS IN STAGE 3 ? 1
STAGE
          NOZZLE WIDTH IN MM? .254
          SUPPLY PRESSURE =
STAGE
                             15.50003100006
                                             TORR
STAGE
          SUPPLY FLOW = .1587499999994 LPM
          SUPPLY RESISTANCE = 97.63799055193 TORR/LPM
STAGE
FIRST STAGE BIAS PRESSURE IN TORR? O
ENTER OUTPUT LOAD IN TERMS OF AN EQUIVALENT LPA
A BLOCKED LOAD IS PUT IN AS AN AR=1 AND BS=0
LOAD NO. UF PARALLEL STAGES? 1
LOAD ASPECT RATIO? 1
LOAD NOZZLE WIDTH(MM)? O
STAGE
          PRESSURE GAIN =
                           7.499062617173
          PRESSURE GAIN = 6.249218847644
STAGE
STAGE
          PRESSURE GAIN =
                           8.333333333333
PRESSURE GAIN OF THE 3 STAGE GAINBLOCK = 390.5273620575
GAINBLOCK INPUT RESISTANCE = 18.30712322849 TORR/LPM
GAINGLOCK OUTPUT RESISTANCE = 48.81899527596 TORR/LPM
```

THIS PROGRAM GIVES DESIGN APPROXIMATIONS FOR PULTISTAGE LPAS BASED ON THE STANDARD HDL LPA, ASSUMING A ZERO-6345, BLOCKED

STOP

Example A-5. Three-stage, parallel-element, common supply gain block.

GAINBLOCK BANDWIDTH AT 90DEG PHASE SHIFT = 813.7516275003 HERTZ

DO YOU WANT TO MAKE ANOTHER CALCULATION? N

```
SINGLE STAGE GAIN OF 10 OPERATING AT SIGMA*NR=1000 IN AIR.
FOR MAXIMUM DYNAMIC RANGE THE PRODUCT OF ASPECT RATIO AND
NOZZLE WIDTH OF EACH SUCEEDING STAGE SHOULD BE 1/3RD OF THE
PREVIOUS, E.G.FOR BS1=BS2, AR(1)=3 AND AR(2)=1.
IS YOUR LPA GAIN 10 ? Y
N' MBER OF STAGES? 3
STAGE
       1 ASPECT RATIO? 3.6
NO. OF PARALLEL ELEMENTS IN STAGE 1 ? 1
          NOZZLE WIDTH IN MM? .508
STAGE
STAGE
           SUPPLY PRESSURE = .2989975115753 TORR
          SUPPLY FLOW = .2160680529282 LPM
SUPPLY RESISTANCE = 1.383811755247 TORR/LPM
STAGE
STAGE
          ASPECT RATIO? 1.2
STAGE
NO. OF PARALLEL ELEMENTS IN STAGE 2 ? 1
STAGE
          NOZZLE WIDTH IN MM? .508
STAGE
           SUPPLY PRESSURE = 2.690977604177
          SUPPLY FLOW = .3148760330576 LPM
STAGE
          SUPPLY RESISTANCE = 8.546149346606 TORR/LPM
STAGE
          ASPECT RATIO? .4
STAGE
NO. OF PARALLEL ELEMENTS IN STAGE 3 ? 1
STAGE
          NOZZLE WIDTH IN MM? .508
          SUPPLY PRESSURE = 24.2187984376 TORR
STAGE
          SUPPLY FLOW = .2991836734683 LPM
STAGE
          SUPPLY RESISTANCE = 93.44260814546 TORR/LPM
FIRST STAGE BIAS PRESSURE IN TORR? O
ENTER OUTPUT LOAD IN TERMS OF AN EQUIVALENT LPA A BLOCKED LOAD IS PUT IN AS AN AR=1 AND BS=0
LOAD NO. OF PARALLEL STAGES? 1
LOAD ASPECT RATIO? 1
```

THIS PROGRAM GIVES DESIGN APPROXIMATIONS FOR MULTISTAGE LPAS BASED ON THE STANDARD HDL LPA, ASSUMING A ZERO-BIAS, BLOCKED

STOP

STAGE

LOAD NOZZLE WIDTH(MM)? O

Example A-6. Three-stage maximum dynamic range gain block.

GAINBLOCK BANDWIDTH AT 90DEG PHASE SHIFT = 117.3680231977 HERTZ

PRESSURE GAIN = 9.025253305181 PRESSURE GAIN = 9.220151573197

PRESSURE GAIN = 9.782608695654

DO YOU WANT TO MAKE ANOTHER CALCULATION? N

PRESSURE GAIN OF THE 3 STAGE GAINBLOCK = 814.0519903724
GAINBLOCK INPUT RESISTANCE = 1.037858816435 TORR/LPM
GAINBLOCK OUTPUT RESISTANCE = 46.72130407273 TORR/LPM

FOR MAXIMUM DYNAMIC RANGE THE PRODUCT OF ASPECT RATIO AND NOZZLE WIDTH OF EACH SUCEEDING STAGE SHOULD BE 1/3RD OF THE PREVIOUS, E.G.FOR BS1=BS2, AR(1)=3 AND AR(2)=1.

```
IS YOUR LPA GAIN 10 ? Y
NUMBER OF STAGES? 3
STAGE 1 ASPECT RATIO? .67
NO. OF PARALLEL ELEMENTS IN STAGE 1 ? 1
           NOZZLE WIDTH IN MM? .76
           SUPPLY PRESSURE = 3.85676529084
STAGE 1
STAGE
           SUPPLY FLOW = .4564523647292 LPM
STAGE
           SUPPLY RESISTANCE = 8.449436543347 TORR/LPM
           ASPECT RATIO? 1
STAGE
NO. OF PARALLEL ELEMENTS IN STAGE 2 ? 1
STAGE
          NUZZLE WIDTH IN MM? .508
STAGE
           SUPPLY PRESSURE = 3.875007750016 TORR
STAGE
           SUPPLY FLOW = .3174999999988 LPM
          SUPPLY RESISTANCE = 12.20474881899 TORR/LPM
STAGE
          ASPECT RATIO? .8
STAGE
NO. OF PARALLEL ELEMENTS IN STAGE 3 ? 1
          NOZZLE WIDTH IN MM? .254
STAGE
       3
          SUPPLY PRESSURE = 24.2187984376
STAGE
          SUPPLY FLOW = .1567901234562 LFM
STAGE
          SUPPLY RESISTANCE = 154.4663522404 TORR/LPM
STAGE
FIRST STAGE BIAS PRESSURE IN TORR? O
ENTER OUTPUT LOAD IN TERMS OF AN EQUIVALENT LPA
A BLOCKED LOAD IS PUT IN AS AN AR=1 AND BS=0
LOAD NO. OF PARALLEL STAGES? 1
LOAD ASPECT RATIO? 1
LOAD NOZZLE WIDTH(MM)? O
STAGE
          PRESSURE GAIN = 6.841026366496
          PRESSURE GAIN = 7.922358876867
PRESSURE GAIN = 9.68992248062
STAGE
STAGE
PRESSURE GAIN OF THE 3 STAGE GAINBLOCK = 525.1653678439
GAINBLOCK INPUT RESISTANCE = 6.33707740751 TORR/LPM
                                77.2331761202 TORR/LPM
GAINGLOCK OUTPUT RESISTANCE =
GAINBLOCK BANDWIDTH AT 90DEG PHASE SHIFT = 226.0758627341 HERTZ DO YOU WANT TO MAKE ANOTHER CALCULATION? NO
```

STOP

The same of the same of the same

Example A-7. Moderately pressure staged, three-stage gain block with change in size.

```
THIS PROGRAM GIVES DESIGN APPROXIMATIONS FOR MULTISTAGE LPAS BASED ON THE STANDARD HDL LPA, ASSUMING A ZERO-BIAS, BLOCKED SINGLE STAGE GAIN OF 10 OPERATING AT SIGMA*NR=1000 IN AIR.

FOR MAXIMUM DYNAMIC RANGE THE PRODUCT OF ASPECT RATIO AND
```

FOR MAXIMUM DYNAMIC RANGE THE PRODUCT OF ASPECT RATIO AND NOZZLE WIDTH OF EACH SUCCEDING STAGE SHOULD BE 1/3RD OF THE PREVIOUS, E.G.FOR BS1=BS2,AR(1)=3 AND AR(2)=1.

```
IS YOUR LPA GAIN 10 ? N
WHAT IS YOUR GAIN? 8
NUMBER OF STAGES? 3
STAGE
          ASPECT RATIO? . 66667
NO. OF PARALLEL ELEMENTS IN STAGE
STAGE
          NOZZLE WIDTH IN MM? .76
STAGE
          SUPPLY PRESSURE = 3.895390408859
STAGE
          SUPPLY FLOW = .9120009119738 LPM
STAGE
          SUPPLY RESISTANCE = 4.271257142088 TORR/LPM
          ASPECT RATIO? .5
STAGE
NO. OF PARALLEL ELEMENTS IN STAGE 2 ? 3
STAGE
          NOZZLE WIDTH IN MM? .508
          SUPPLY PRESSURE = 15.50003100006
STAGE
STAGE
          SUPPLY FLOW = .846666666633 LPM
STAGE
          SUPPLY RESISTANCE = 18.30712322849 TORR/LPM
          ASPECT RATIO? .3333
STAGE
NO. OF PARALLEL ELEMENTS IN STAGE 3 ? 6
          NOZZLE WIDTH IN MM? .375
STAGE
          SUPPLY PRESSURE = 64.01280192128
                                             TORR
STAGE
          SUPPLY FLOW = 1.054634762344 LPM
STACE
          SUPPLY RESISTANCE = 60.69665462099 TORR/LPM
FIRST STAGE BIAS PRESSURE IN TORR? O
ENTER OUTPUT LOAD IN TERMS OF AN EQUIVALENT LPA
A BLOCKED LOAD IS PUT IN AS AN AR=1 AND BS=0
LOAD NO. OF PARALLEL STAGES? 1
LOAD ASPECT RATIO? 1
LOAD NOZZLE WIDTH(MM)? O
STAGE
         PRESSURE GAIN =
                          6.922699358662
         PRESSURE GAIN = 6.341389575126
STAGE
STAGE
         PRESSURE GAIN =
                           7.630472181181
PRESSURE GAIN OF THE 3 STAGE GAINBLOCK = 334.97416948
GAINBLOCK INPUT RESISTANCE = 3.203442856566 TORR/LPM
GAINBLOCK OUTPUT RESISTANCE = 30.34832731049 TORR/LFM
GAINBLOCK BANDWIDTH AT 90DEG PHASE SHIFT = 280.7632244981 HERTZ
DO YOU WANT TO MAKE ANOTHER CALCULATION? N
```

STOP

Example A-8. Flueric servovalve gain block.

```
NUMBER OF STAGES? 8
           ASPECT RATIO? 1.07
STAGE
NO. OF PARALLEL ELEMENTS IN STAGE
STAGE
           NOZZLE WIDTH IN MM? .76
           SUPPLY PRESSURE = 1.512186163908
STAGE
STAGE
           SUPPLY FLOW = _4744568134623 LPM
           SUPPLY RESISTANCE = 3.187194536997 TGRR/LPM
STAGE
STAGE
           ASPECT RATIO? 1
NO. OF PARALLEL ELEMENTS IN STAGE
                                     2 ? 1
STAGE
           NOZZLE WIDTH IN MM? .76
STAGE
           SUPPLY PRESSURE = 1.731301939058
           SUPPLY FLOW = .4750000000058 LPM
STAGE
           SUPPLY RESISTANCE = 3.644846187446 TORR/LPM
STAGE
STAGE
           ASPECT RATIO? .833
NO. OF PARALLEL ELEMENTS IN STAGE
                                     3 ? 1
STAGE
          NOZZLE WIDTH IN MM? .76
           SUPPLY PRESSURE = 2.495070449392
                                               TCRR
STAGE
STAGE
           SUPPLY FLOW = .4710572283805 LPM
          SUPPLY RESISTANCE = 5.296745913379 TORR/LPM
STAGE
           ASPECT RATIO? .66667
STAGE
NO. OF PARALLEL ELEMENTS IN STAGE
STAGE
          NOZZLE WIDTH IN MM? .76
STAGE
          SUPPLY PRESSURE = 3.895390408859
          SUPPLY FLOW = .4560004559869 LPM
STAGE
          SUPPLY RESISTANCE = 8.542514284176 TORR/LPM
STAGE
          ASPECT RATIO? .5333
STAGE
NO. OF PARALLEL ELEMENTS IN STAGE 5 ? 1
STAGE
          NOZZLE WIDTH IN MM? .76
STAGE
          SUPPLY PRESSURE = 6.08736927703
          SUPPLY FLOW = .4307936914142 LPM
STAGE
STAGE
          SUPPLY RESISTANCE = 14.12403336359 TORR/LPM
STAGE 6 ASPECT RATIO? .333
NO. OF PARALLEL ELEMENTS IN STAGE 6 ? 1
STAGE
          NOZZLE WIDTH IN MM? .76
          SUPPLY PRESSURE = 15.61292769398
STAGE
                                               TORR
STAGE
          SUPPLY FLOW = .356071763629 LPM
STAGE
          SUPPLY RESISTANCE = 43.84769950545 TORR/LPM
          ASPECT RATIO? .45
STAGE
NO. OF PARALLEL ELEMENTS IN STAGE
STAGE
          NOZZLE WIDTH IN MM? .508
STAGE
          SUPPLY PRESSURE = 19.13584074082
STAGE
          SUPPLY FLOW = .271819262763 LPM
          SUPPLY RESISTANCE = 70.39913413901 TORR/LPM
STAGE
STAGE
          ASPECT RATIO? .3
NO. OF PARALLEL ELEMENTS IN STAGE
                                    8 ? 1
STAGE
          NOZZLE WIDTH IN MM? .76
STAGE
          SUPPLY PRESSURE = 19.23668821176
          SUPPLY FLOW = .3372781065064 LPM
STAGE
STAGE
       8
          SUPPLY RESISTANCE = 57.0350931195 TORR/LPM
 Example A-9. Eight-stage M48A5 gun stabilization system gain block
             (cont'd).
```

```
FIRST STAGE BIAS PRESSURE IN TORR? 0
ENTER OUTPUT LOAD IN TERMS OF AN EQUIVALENT LPA
A BLOCKED LOAD IS PUT IN AS AN AR=1 AND BS=0
LOAD NO. OF PARALLEL STAGES? 1
LOAD ASPECT RATIO? 1
LOAD NOZZLE WIDTH(HM)? 0
STAGE
           PRESSURE GAIN =
                               6.316119474231
STAGE
           PRESSURE GAIN =
                               5.834821179151
        3
STAGE
           PRESSURE CAIN =
                               6.212174122533
           PRESSURE GAIN = PRESSURE GAIN =
STAGE
                               6.316327274507
STAGE
                               7.297506173708
           PRESSURE JAIN =
                               6.553892007645
STAGE
STAGE
           PRESSURE GAIN =
                               4.715178897435
STAGE
       8
           PRESSURE GAIN =
                              8.340620888182
PRESSURE GAIN OF THE
                        8 STAGE GAINBLOCK = 2719921.938609
GAINBLOCK INPUT RESISTANCE = 2.390395902748 TORR/LPM
GAINBLOCK OUTPUT RESISTANCE = 28.51754655975 TORR/LPM
GAINBLOCK BANDWIDTH AT 90DEG PHASE SHIFT = 55.2316647611 HERTZ
DO YOU WANT TO MAKE ANOTHER CALCULATION? NO
```

Example A-9. Eight-stage M48A5 gun stabilization system gain block (cont'd).

STOP

APPENDIX B.--REPRESENTATIVE STACKING ORDERS

This appendix presents the detailed stacking orders for three specific gain block configurations.

EXAMPLE B-1. STACKING ORDER FOR TWO-STAGE, SELF-STAGED GAIN BLOCK

Part	Orientation	Notes
5035	λ .	Transfer control signal to outer holes
5018a	H	Blocks control signal from second stage
Vent	H	5011a or 5239, depending on b
Kidney	, A	5215a, 5237a, 5242a, or 5340a
Butterfly	` A	5137a, 5236a, 5241a, or 5339a
LPAd	A	630xx, 620xx, 615xx, or other
Butterfly	A	••
Kidney	Α	••
Vent	н	
5022a	D	Keeps vents from communicating
5018a	A	Blocks first-stage controls from second stage
Vent	н	•
Kidney	H	••
Butterfly	н	
LPA	H	••
Butterfly	H	••
Kidney	н	
Vent	н ,	••
5040a	A	••
5117	H	Transfer control signal to first-stage inputs
5040a	. A 🕽	
5011a	E (Paralleled supply transfer for quietness.
5040a	A C	rarariora adhit pranarar tar daracuess
5011a	E	
Cover plat	c	••

^aLPA: laminar proportional amplifier

APPENDIX B

EXAMPLE B-2. GAIN BLOCK OF FLUIDIC CAPILLARY PYROMETER

Part	Orientation	Notes
043	**	Not needed when base is "O" ringed
005	~-	Not needed with good filtration
5109	F	
5116	H	
5111	G	
5221a	A	5 mils thick
5018a	A	
3221a	A	
018a	Α \	
011a	λ }	
339a	c /	
1505	c }	Second-stage LPA, a aspect ratio = 0.33
339a	c (•
011a	λ }	
018a	н /	
011a	Αı	
339a	F	
1510	A (First-stage LPA, aspect ratio = 0.67
339a	F (
011a	Α)	
018a	A /	
118a	λ	
018A	н	
114	Н	
039	F	
112	В	

^aLPA: laminar proportional amplifier

EXAMPLE B-3. THREE-STAGE MAXIMUM DYNAMIC RANGE GAIN BLOCK

Part	Orientation	Notes
5239a(2)		Exhaust
5215a	C	Vent collector
5137a	Ċ	Vent plate
62012(6)	С	First-stage laminar proportional amplifier, aspect ratio = 3.6
5137a	C	
5215a	С	••
5239a(2)	A	•-
5022a	В	Isolate first-stage venting
5188a(3)	C .	Transfer input signal into first stage
5018a	A)	
5011a	D }	Three pairs, first-stage supply transfer
5021a	A ´	Block main pressure from first stage
5221a (4 mils) A)	Four pairs, first-stage dropping resistor
5018a	A }	rour parts, ritst-stage dropping resistor
5221a (4 mils) A ´	••
5021a	C	<pre>Block second-stage pressure from first stage</pre>
5239a	A	••
5215a	A	••
5137a	λ	
62012(2)	A	Second-stage laminar proportional
		amplifier, aspect ratio = 1.2
5137a	A	
5215a	A	••
5018a	С	Block second stage controls, transfer to third stage
5239a	A	••
5215a	C	
5137a	· - · C	
62008	C .	••
5137a	С	
5215a	С	
5239a	A	
5018a	A }	Five pairs, second-stage dropping
5221a (5 mils)) A }	resistor and third-stage control blockage
5118a(3)	C	Transfer input signal to outer holes to transfer to first stage

DISTRIBUTION

ADMINISTRATOR
DEFENSE TECHNICAL INFORMATION CENTER
ATTN DTIC-DDA (12 COPIES)
CAMERON STATION, BUILDING 5
ALEXANDRIA, VA 22314

COMMANDER
US ARMY RSCH & STD GP (EUR)
ATTN CHIEF, PHYSICS & MATH BRANCH
FPO NEW YORK 09510

COMMANDER
US ARMY MATERIEL DEVELOPMENT &
READINESS COMMAND
ATIN DRCLDC, JAMES BENDER
5001 EISENHOWER AVENUE
ALEXANDRIA, VA 22333

COMMANDER
US ARMY ARMAMENT MATERIEL
READINESS COMMAND
ATTN DRSAR-ASF, FUZE 6
MUNITIONS SUPPORT DIV
ATTN DRSAR-RDF, SYS DEV DIV-FUZES
ATTN DRSAR-RDG-T, R. SPENCER
ATTN DRSAR-ASF
ATTN DRSAR-LEP-L, TECH LIBRARY
ROCK ISLAND, IL 61299

COMMANDER
US ARMY MISSILE & MUNITIONS
CENTER & SCHOOL
ATTN ATSK-CTD-F
REDSTONE ARSENAL, AL 35809

DIRECTOR
US ARMY MATERIEL SYSTEMS
ANALYSIS ACTIVITY
ATTN DRXSY-MP
ABERDEEN PROVING GROUND, MD 21005

DIRECTOR
US ARMY BALLISTIC RESEARCH LABORATORY
ATTN DRDAR-TSB-S (STINFO)
ABERDEEN PROVING GROUND, MD 21005

US ARMY ELECTRONICS TECHNOLOGY 6 DEVICES LABORATORY ATTN DELET-DD FT MONMOUTH, NJ 07703

HQ, USAF/SAMI WASHINGTON, DC 20330

TELEDYNE BROWN ENGINEERING CUMMINGS RESEARCH PARK ATTN MELVIN L. PRICE, MS~44 HUNTSVILLE, AL 35807 ENGINEERING SOCIETIES LIBRARY ATTN ACQUISITIONS DEPARTMENT 345 EAST 47TH STREET NEW YORK, NY 10017

COMMANDER IDDR&E
PENTAGON, ROOM 3D 1089
ATTN G. KOPCSAK
WASHINGTON, DC 20310

OFFICE OF THE DEPUTY CHIEF OF STAFF FOR RESEARCH, DEVELOPMENT & ACQUISITION DEPARTMENT OF THE ARMY ATTN DAMA-ARP-P ATTN DAMA-CSS-N WASHINGTON, DC 20310

US ARMY R&D GROUP (EUROPE) BOX 15 ATTN CHIEF, AERONAUTICS BRANCH ATTN CHIEF, ENGINEERING SCIENCES FPO NEW YORK 09510

US ARMY RESEARCH OFFICE PO BOX 12211 ATTN R. SINGLETON RESEARCH TRIANGLE PARK, NC 27709

BMD ADVANCED TECHNOLOGY CENTER PO BOX 1500 ATTN J. PAPADOPOULOS HUNTSVILLE, AL 35807

COMMANDER
US ARMY FOREIGN SCIENCE
6 TECHNOLOGY CENTER
FEDERAL OFFICE BUILDING
ATTN DRXST-SD1
ATTN DRXST-IS3, C. R. MOORE
220 7TH STREET, NE
CHARLOTTESVILLE, VA 22901

DIRECTOR
APPLIED TECHNOLOGY LABORATORY
ATTN GEORGE W. FOSDICK, DAVDL-ATL-ASA
FT EUSTIS, VA 23604

COMMANDER
US ARMY MATERIEL AND MECHANICS
RESEARCH CENTER
ATTN R. KATZ
WATERTOWN, MA 02172

COMMANDER
USA MISSILE COMMAND
ATTN REDSTONE SCIENTIFIC INFORMATION
CENTER, DRSMI-RBD
ATTN DRDMI-TGC, WILLIAM GRIFFITH

USA MISSILE COMMAND (Cont'd) ATTN DRDMI-TGC, '. C. DUNAWAY ATTN DRCPM-TOE, FRED J. CHEPLEN REDSTONE ARSENAL, AL 35898

COMMANDER
US ARMY MOBILITY EQUIPMENT RED CENTER
ATTN TECHNICAL LIBRARY (VAULT)
ATTN DRDME-EM, R. N. WARE
FT BELVOIR, VA 22060

COMMANDER
EDGEWOOD ARSENAL
ATTN SAREA-MT-T, D. PATTON
ABERDEEN PROVING GROUND, MD 21010

COMMANDER
US ARMY ARRADCOM
ATTN SARPA-TS-S #59
ATTN DRDAR-LCN-C, A. E. SCHMIDLIN
ATTN DRDAR-LCW-E, J. CONNORS
ATTN DRDAR-SCF-IC, V. BAUMCARTH
ATTN PBM-DPM (TAGLAIRINO)
ATTN PBM-MG (A. WILLIAMS)
DOVER, NJ 07801

COMMANDER
WATERVLIET ARSENAL
ATTN SARWV-RDT-L
ATTN DRDAR-LCB-RA, R. RACICOT
WATERVLIET ARSENAL, NY 12189

COMMANDER
US ARMY TANK AUTOMOTIVE RES &
DEV COMMAND
ARMOR & COMP DIV, DRDTA-RKT
BLDG. 215
ATTN M. WHITMORE
WARREN, MI 48090

COMMANDER ATTN STEWS-AD-L, TECHNICAL LIBRARY WHITE SANDS MISSILE RANGE, NM 88002

COMMANDER/DIRECTOR
ATMOSPHERIC SCIENCES LABORATORY
USA ERADCOM
ATTN DELAS-AS (HOLT)
ATTN DELAS-AS-T (R. RUBIO)
WHITE SANDS MISSILE RANGE, NM 88002

OFFICE OF NAVAL RESEARCH DEPARTMENT OF THE NAVY ATTN STANLEY W. DOROFF, CODE 438 ATTN D. S. SIEGEL, CODE 211 ARLINGTON, VA 22217

DEPARTMENT OF THE NAVY
R&D PLANS DIVISION
ROOM 5D760, PENTAGON
ATTN BENJ R. PETRIE, JR.
OP-987P4
WASHINGTON, DC 20350

COMMANDANT
US NAVAL POSTGRADUATE SCHOOL
DEPARTMENT OF MECHANICAL ENGINEERING
ATTN CODE 69 Nn(NUNN)
MONTEREY, CA 93940

COMMANDER
NAVAL AIR DEVELOPMENT CENTER
ATTN R. MCGIBONEY, 60134
ATTN CODE 8134, LOIS GUISE
ATTN D. KEYSER, 60134
WARMINSTER, PA 18974

COMMANDER OFFICER
NAVAL AIR ENGINEERING CENTER
ATTN ESSD, CODE 9314, HAROLD OTT
LAKEHURST, NY 08733

NAVAL AIR SYSTEMS COMMAND
DEPARTMENT OF THE NAVY
ATTN CODE AIR-5162C1, J. BURNS
ATTN CODE AIR-5143J, D. HOUCK
WASHINGTON, DC 20361

COMMANDER
PACIFIC MISSILE TEST CENTER
ATTN CODE 3123, ABE J. GARRETT
ATTN CODE 1243, A. ANDERSON
POINT MUGU, CA 93042

COMMANDER
NAVAL SHIP ENGINEERING CENTER
PHILADELPHIA DIVISION
ATTN CODE 6772
PHILADELPHIA, PA 19112

COMMANDER
NAVAL SURFACE WEAPONS CENTER
ATTN CODE 413, CLAYTON MCKINDRA
WHITE OAK, MD 20910

COMMANDER
NAVAL ORDNANCE STATION
ATTN CODE 5123C, K. ENGLANDER
INDIANHEAD, MD 20640

NAVAL SHIP RES & DEV CENTER CODE 1619, K. READER BETHESDA, MD 20084

NAVAL RESEARCH LABORATORY ATTN 5. SEARLES, 117 BG A68 WASHINGTON, DC 20375

NAVAL SEA SYSTEMS COMMAND SEA05R11 ATTN J. H. HAPRISON WASHINGTON, DC 20362

COMMANDER
NAVAL WEAPONS CENTER
ATTN CODE 533, LIBRARY DIVISION
ATTN CODE 3636, C. BURMEISTER
CHINA LAKE, CA 93555

COMMANDER
AF AERO PROPULSION LABORATORY, AFSC
ATTN LESTER SMALL, AFWAL/POTC
WRIGHT-PATTERSON AFB, OH 45433

COMMANDER
AIR FORCE AVIONICS LABORATORY
ATTN AARA-2, RICHARD JACOBS
WRIGHT-PATTERSON AFB, OH 45433

DIRECTOR
AF OFFICE OF SCIENTIFIC RESEARCH
ATTN NE
BOLLING AFB, DC 20332

COMMANDER
AIR FORCE FLIGHT DYNAMICS LABORATORY
ATTN AFFDL/FGL, H. SNOWBALL
ATTN AFFDL/FER, R. J. DOBBEK
WRIGHT-PATTERSON AFB, OH 45433

COMMANDER
AF WEAPONS LABORATORY, AFSC
ATTN SUL, TECHNICAL LIBRARY
KIRTLAND AFB, NM 87117

COMMANDER
ARMAMENT DEVELOPMENT AND TEST CENTER
ATTN ADTC (DLOSL), TECH LIBRARY
ATTN DLMA, DAVID T. WILLIAMS
EGLIN AIR FORCE BASE, FL 32542

AIR FORCE FLIGHT TEST CENTER 6510 ABG/SSD ATTN TECHNICAL LIBRARY EDWARDS AFB, CA 93523

AF INSTITUTE OF TECHNOLOGY, AU
ATTN LIBRARY AFIT (LD),
BLDG 640, AREA B
ATTN AFIT (ENM), MILTON E. FRANKE
WRIGHT-PATTERSON AFB, OH 45433

HQ, AF SYSTEMS COMMAND ATTN SGB, CPT GEORGE JAMES ANDREWS AFB, DC 20334 ARGONNE NATIONAL LABORATORY
APPLIED PHYSICS DIV, BLDG 316
ATTN N. M. O'FALLON
9700 S. CASS AVE
ARGONNE, IL 60439

OAK RIDGE NATIONAL JABORATORY
CENTRAL RES LIBRARY, BLDG 4500N, RM 175
PO BOX X
ATTN E. HOWARD
ATTN C. A. MOSSHAN
ATTN R. E. HARPER
ONK RIDGE, TN 37830

DEPARTMENT OF COMMERCE
NATIONAL BUREAU OF STANDARDS
ATTN JAMES SCHOOLEY, CHIEF,
TEMPERATURE SECTION
ATTN T. NEGAS, SOLID STATE
CHEMISTRY DIVISION
ATTN RAY DILS, RM B-254, BLDG 221
ATTN GEORGE BURNS, RM B-222, BLDG 221
WASHINGTON, DC 20234

DEPARTMENT OF COMMERCE BUREAU OF EAST-WEST TRADE OFFICE OF EXPORT ADMINISTRATION ATTN WALTER J. RUSNACK WASHINGTON. DC 20230

DEPARTMENT OF ENERGY C-156, GTN (OART) ATTN ROBERT ROBERTS ATTN SANDY DAPKUNAS WASHINGTON, DC 20545

DEPARTMENT OF ENERGY FE-22 ATTN T. K. LAU WASHINGTON, DC 20545

DEPARTMENT OF ENERGY F-317, GTN (COAL GASIFICATION) ATTN JIM CARR WASHINGTON, DC 20545

FEDERAL BUREAU OF INVESTIGATION
J. EDGAR HOOVER BLDG
ATIN ROBERT WILLIS
WASHINGTON, DC 20535

DEPARTMENT OF JUSTICE
IMMIGRATION AND NATURALIZATION SERVICE
425 "1" STREET, NW
ATTN NEILL MCKAY
WASHINGTON, DC 20536

SCIENTIFIC LIBRARY
US PATENT OFFICE
ATTN MRS. CURETON
WASHINGTON, DC 20231

NASA AMES RESEARCH CENTER ATTN MS 244-13, DEAN CHISEL MOFFETT FIELD, CA 94035

NASA LANGLEY RESEARCH CENTER
ATTN MS 494, H. D. GARNER
ATTN MS 494, R. R. HELLBAUM
ATTN MS 185, TECHNICAL LIBRARY
HAMPTON, VA 23665

NASA SCIENTIFIC & TECH INFO FACILITY PO BOX 8657 ATTN ACQUISITIONS BRANCH BALTIMORE/WASHINGTON INTERNATIONAL AIRPORT, MD 21240

UNIVERSITY OF ALABAMA
CIVIL & MINERAL ENGINEERING DEPT
PO BOX 1468
ATTN HAROLD R. HENRY
UNIVERSITY, AL 35486

UNIVERSITY OF ARKANSAS TECHNOLOGY CAMPUS PO BOX 3017 ATTN PAUL C. MCLEOD LITTLE ROCK, AR 72203

UNIVERSITY OF ARKANSAS
MECHANICAL ENGINEERING
ATTN JACK H. COLE, ASSOC. PROF.
FAYETTEVILLE, AR 72701

CARNEGIE-MELLON UNIVERSITY SCHENLEY PARK ATTN PROF. W. T. ROULEAU, MECH ENGR DEPT PITTSBURGH, PA 15213

CASE WESTERN RESERVE UNIVERSITY ATTN PROF. P. A. ORNER ATTN PROF. B. HORTON UNIVERSITY CIRCLE CLEVELAND, OH 44106

THE CITY COLLEGE OF THE CITY
UNIVERSITY OF NY
DEPT OF MECH ENGR
ATTN PROF. L. JIJI
ATTN PROF. G. LOWEN
139TH ST. AT CONVENT AVE
NEW YORK, NY 10031

CLEVELAND STATE UNIVERSITY FENN COLLEGE OF ENGINEERING ATTN PRCF. R. COMPARIN CLEVELAND, OH 44115

DUKE UNIVERSITY
COLLEGE OF ENGINEERING
ATTN C. M. HARMAN
DURHAM, NC 27706

ENGINEERING SOCIETIES LIBRARY ATTN HOWARD GORDON ATTN ACQUISTRIONS DEPARTMENT 345 EAST 47T. DET NEW YORK, NY 10017

FPANKLIN INSTITUTE OF THE STATE

OF PENNSYLVANIA

ATTN KA-CHEUNG TSUI, ELEC ENGR DIV

ATTN C. A. BELSTERLING

20TH STREET & PARKWAY

PHILADELPHIA, PA 19103

HUGHES HELICOPTERS
DIVISION OF SUMMA CORPORATION
CENTINELA & TEALE STREETS
ATTN LIBRARY 2/T2124
CULVER CITY, CA 90230

IIT RESEARCH INSTITUTE ATTN K. E. MCKEE 10 WEST 35TH STREET CHICAGO, IL 60616

JET PROPULSION LABORATORY ATTN JOHN V. WALSH, MS 125-138 4800 OAK GROVE DRIVE PASADENA, CA 91103

JOHNS HOPKINS UNIVERSITY
APPLIED PHYSICS LABORATORIES
ATTN MAYNARD HILL
ATTN THOMAS RANKIN
ATTN JOSEPH WALL
LAUREL, MD 20810

LEHIGH UNIVERSITY
DEPARTMENT OF MECHANICAL ENGINEERING
ATTN PROF. FORBES T. BROWN
BETHLEHEM, PA 18015

LINDA HALL LIBRARY ATTN DOCUMENTS DIVISION 5109 CHERRY STREET KANSAS CITY, MO 64110

LOS ALAMOS SCIENTIFIC LAB PO BOX 1663 ATTN FRANK FINCH, MS 178 LOS ALAMOS, NM 87545

MASSACHUSETTS INSTITUTE OF TECHNOLOGY ATAN ENGINEERING TECHNICAL REPORTS, RM 10-408 ATTN DAVID WORMLEY, MECH ENGR DEPT, RM 3-146 77 MASSACHUSETTS AVENUE CAMBRIDGE, MA 02139

MICHIGAN TECHNOLOGICAL UNIVERSITY LIBRARY, DOCUMENTS DIVISION ATTN J. HAWTHORNE HOUGHTON, MI 49931

UNIVERSITY OF MISSISSIPPI ATTN JOHN A. FOX 201 CARRIER HALL, DEPT OF MECH ENGR UNIVERSITY, MS 38677

MISSISSIPPI STATE UNIVEPSITY DRAWER ME ATTN C. J. BELL, MECH ENG DEPT STATE COLLEGE, MS 39762

MISSISSIPPI STATE UNIVERSITY
DEPT OF AEROSPACE ENGINEERING
ATTN CAVID MURPHREE
MISSISSIPPI STATE, MS 39762
UNIVERSITY OF NEBRASKA LIBRARIES
ACQUISITIONS DEPT, SERIALS SECTIONS
ATTN ALAN GOULD
LINCOLN, NE 68508

UNIVERSITY OF NEW HAMPSHIRE MECH ENGR DEPT, KINGSBURY HALL ATTN PROF. CHARLES TAFT ATTN PROF. DAVID LIMBERT DURHAM, NH 03824

UNIVERSITY OF N CAROLINA
INSTITUTE OF MARINE BIOMEDICAL RESEARCH
ATTN MICHAEL E. SHEEHAN
WILMINGTON, NC 28401

NEW JERSEY INSTITUTE OF TECHNOLOGY
DEPARTMENT OF MECHANICAL ENGINEERING
ATTN R. Y. CHEN
323 HIGH STREET
NEWARK, NJ 07102

OHIO STATE UNIVERSITY LIBRARIES SERIAL DIVICION, MAIN LIBRARY 1858 NEIL AVENUE COLUMBUS, OH 43210

OKLAHOMA STATE UNIVERSITY SCHOOL OF MECH & AEROSPACE ENGR ATTN PROF. KARL N. REID STILLWATER, OK 74074 MIAMI UNIVERSITY
DEPT OF ENG TECH
SCHOOL OF APPLIED SCIENCE
ATTN PROF. S. B. FRIEDMAN
OXFORD, OH 45056

PENNSYLVANIA STATE UNIVERSITY ATTN J. L. SHEARER 215 MECHANICAL ENGINEERING BUILDING UNIVERSITY PARK, PA 16802

PENNSYLVANIA STATE UNIVERSITY ENGINEERING LIBRARY ATTN M. BENNETT, ENGINEFRING LI'RARIAN 201 HAMMOND BLDG UNIVERSITY PARK, PA 16802

PORTLAND STATE UNIVERSITY
DEPT OF ENGINEERING AND
APPLIED SCIENCE
PO BOX 751
ATTN PROF. P. I. CHEN
PORTLAND, OR 97207

PURDUE UNIVERSITY
SCHOOL OF MECHANICAL ENGINEERING
ATTN PROF.. VICTOR W. GOLDSCHMIDT
ATTN PROF.. ALAN T. MCDONALD
LAFAYETTE, IN 47907

ROCK VALLEY COLLEGE ATTN KEN B! 3TON 3301 N. MULFORD ROAD ROCKFORD, IL 61101

RUTGERS UNIVERSITY
LIBRARY OF SCIENCE & MEDICINE
ATTN GOVERMENT DOCUMENTS DEPT
SANDRA R. LIVINGSTON
NEW BRUNGWICK, NJ 08903
SYRACUSE UNIVERSITY
DEPT OF MECH & APROSPACE ENGINEERING
ATTN PROF.ESSOR D. S. DOSANJH
139 L. A. LINK HALL
SYRACUSE, NY 13210

UNIVERSITY OF TENNESSEE
DEPT OF MECHANICAL ENGINEERING
ATTN PROF. G. V. SMITH
KNOXVILLE, TN 37916

UNIVERSITY OF TENNESSEE SPACE INST ENERGY CONVERSION DIVISION ATTN MARY ANN SCOTT TULLAHOMA, TN 37388

UNIVERSITY OF TEXAS AT AUSTIN DEPT OF MECHANICAL ENGINEERING ATTN A. J. HEALEY AUSTIN, TX 78712

THE UNIVERSITY OF TEXAS AT ARLINGTON MECHANICAL ENGINEERING DEPARTMENT ATTN ROBERT L. WOODS ARLINGTON, TX 76019

TULANE UNIVERSITY
DEPT OF MECHANICAL ENGINEERING
AATN H. F. HRUBECKY
NEW ORLEANS, LA 70118

UNION COLLEGE
MECHANICAL ENGINEERING
ATTN ASSOC. PROF. W. C. AUBREY
MECH ENGR DEPT, STEINMETZ HALL
SCHENECTADY, NY 12308

UNIVERSITY OF VIRGINIA
DEPT OF MECH & AEROSPACE ENGR
ATTN DAVID LEWIS
CHARLOTTESVILLE, VA 22090

VIRGINIA POLYTECHNIC INSTITUTE
OF STATE UNIV
MECHANICAL ENGINEERING DEPARTMENT
ATTN PROF. H. MOSES
BLACKSBURG, VA 24061

WASHINGTON UNIVERSITY SCHOOL OF ENGINEERING PO BOX 1185 ATTN W. M. SWANSON ST LOUIS, MO 63130

WEST VIRGINIA UNIVERSITY
MECHANICAL ENGINEERING DEPARTMENT
ATTN RICHARD A. BAJURA
MORGANTOWN, WV 26505

WICHITA STATE UNIVERSITY ATTN DEPT AERO ENGR, E. J. RODGERS WICHITA, KS 67208

UNIVERSITY OF WISCONSIN
MECHANICAL ENGINEERING DEPARTMENT
ATTN FEDERAL REPORTS CENTER
ATTN NORMAN H. HEACHLEY, DIR,
DESIGN ENGINEERING LABORATORIES
1513 UNIVERSITY AVENUE
MADISON, WI 53706

WORCESTER POLYTECHNIC INSTITUTE ATTN GEORGE C. GORDON LIBRARY (TR) ATTN TECHNICAL REPORTS WORCESTER, MA 01609

AVCO SYSTEMS DIVISION ATTN W. K. CLARK 201 LOWELL STREET WILMINGTON, MA 01887

The second second

BARNES ENGINEERING CO ATTN FRED SWEIBAUM 30 COMMERCE ROAD STANFORD, CT 06904

BELL HELICOPTER COMPANY PO BOX 482 ATTN R. D. YEARY FORTWORTH, TX 76101

BENDIX CORPORATION
ELECTRODYNAMICS DIVISION
ATTN D. COOPER
11600 SHERMAN WAY
N. HOLLYWOOD, CA 90605

BENDIX CORPORATION
RESEARCH LABORATORIES DIV
BENDIX CENTER
ATTN C. J. AHERN
ATTN LAEL TAPLIN
SOUTHFIELD, MI 48075

BOEING COMPANY, THE PO BOX 3707 ATTN HENRIK STRAUB SEATTLE, WA 98124

BOWLES FLUIDICS CORPORATION ATTN VICE PRES/ENGR 9347 FRASER AVENUE SILVER SPRING, MD 20910

RONALD BOWLES 2105 SONDRA COURT SILVER SPRING, MD 20904

CHAMBERLAIN MANUFACTURING CORP EAST 4TH & ESTHER STS PO BOX 2545 WATERLOO, IA 50705

CONTINENTAL CAN COMPANY TECH CENTER ATTN P. A. BAUER 1350 W. 76TH STREET CHICAGO, IL 60620

CORDIS CORPORATION
PO BOX 428
ATTN STEPHEN F. VADAS, K-2
MIAMI, FL 33137

CORNING GLASS WORKS FICIDIC PRODUCTS ATTN R. H. BELLMAN HOUGHTON PARK, B-2 CORNING, NY 14830

CHRYSLER CORPORATION PO BOX 118 CIMS-418-33-22 ATTN L. GAU DETROIT, MI 48231

JOHN DEERE PRODUCT ENCINEERING CENTER ATTN V. S. KUMAR WATERLOO, IA 50704

ELECTRIC POWER RESEARCH INSTITUTE
PO BOX 10412
ATTN MS. M. ANGWIN,
P. M. GEOTHERMAL ENERGY
3412 HILLVIEW AVE
PALO ALTO, CA 94303

FLUIDICS QUARTERLY PO BOX 2989 ATTN D. H. TARUMOTO STANFORD, CA 94305

FOXBORO CO
CORPORATE RESEARCH DIV
ATTN JAMES VIGNOS
ATTN J. DECARLO
ATTN JOHN CHANG
38 NEPONSET AVE
FOXBORO, MA 02035

GARRETT PNEUMATIC SYSTEMS DIVISION
PO BOX 5217
ATTN GARY FREDERICK
ATTN TREVOR SUTTON
ATTN TOM TIPPETTS
ATTN C. ABBOTT
111 SOUTH 34TH STREET
PHOENIX, AZ 85010

GENERAL ELECTRIC COMPANY SPACE/RESD DIVISIONS PO BOX 8555 ATTN MGR LIBRARIES, LARRY CHASEN PHILADELPHIA, PA 19101

GENERAL ELECTRIC COMPANY KNOLLS ATOMIC POWER LABORATORY ATTN D. KROMMENHOEK SCHENECTADY, NY 12301

GENERAL MOTORS CORPORATION DELCO ELECTRONICS DIV MANFRED G. WRIGHT NEW COMMERCIAL PRODUCTS PO BOX 1104 ATTN R. E. SPARKS KOKOMO, IN 46901 GRUMMAN AEROSPACE CORPORATION TECHNICAL INFORMATION CENTER ATTN C. W. TURNER, DOCUMENTS LIBRARIAN ATTN TED SORENSEN, MS B1535 ATTN JACK LEONARD, MS B1535 SOUTH OYSTER BAY ROAD BETHPAGE, L. I., NY 11714

HAMILTON STANDARD DIVISION OF UNITED AIRCRAFT CORPORATION ATTN PHILIP BARNES WINDSOR LOCKS, CT 06096

HONEYWELL, INC ATTN J. HEDEEN ATTN W. POSINGIES 1625 ZARTHAN AVE MINNEAPOLIS, MN 55413

HONEYWELL, INC ATTN RICHARD STEWART, MS 200 1100 VIRGINIA DRIVE FT WASHINGTON, PA 19034

JOHNSON CONTROLS, INC ATTN WARREN A. LEDERMAN ATTN GEORGE JANU 507 E. MICHIGAN MILWAUKEE, WI 53201

LEEDS & NORTHRUP CO ATTN ERNEST VAN VALKENBURG DICKERSON ORAD NORTH WALES, PA 19454

MOORE PRODUCTS COMPANY ATTN R. ADAMS SPRING HOUSE, PA 19477

MARTIN MARIETTA CORPORATION AEROSPACE DIVISION ATTN R. K. BRODERSON, MP 326 PO BOX 5837 ORLANDO, FL 32805

MCDONNELL AIRCRAFT COMPANY
GUIDANCE AND CONTROL MECHANICS DIVISION
ATTN LOYAL GUENTHER
ST LOUIS, MO 63166

MCDONNELL DOUGLAS ASTRONAUTICS CO PROPULSION DEPARTMENT ATTN V. E. HALOULAKOS (A3-226) ATTN J. D. SCHWEIKLE (A3-226) 5301 BOLSA AVENUE HUNTINGTON BEACH, CA 92647

NATIONAL FLUID POWER ASSOC. ATTN JOHN R. LUEKE DIR OF TECH SERVICES 3333 NORTH MAYFAIP ROAD MILWAUKEE, WI 53-22

NEOS, INC 3711 AIR PARK RD ATTN A. J. OSTDIEK LINCOLN, NE 68524

NORTHRUP CORP, ELECTRONICS DIV ATTN DESMOND NELSON, SENIOR ENGINEER ORGN C3133, W/C 2301 W. 12CTH ST HAWTHORNE, CA 90250

PATSCENTER INTERNATIONAL 707 ALEXANDER ROAD ATTN MR. JOHN CLINE PRINCETON, NJ 08540

PLESSEY AEROSPACE LTD ATTN A. ROSENBERG 1700 OLD MEADOW ROAD MCLEAN, VA 22102

PROCON, INC ATTN HERB MARCH OUP PLAZA DES PLAINES, IL 60016

PROPULSION DYNAMICS ATTN T. HOULIHAN 2200 SOMERVILLE R ANNAPOLIS, MD 21401

RICHARD WHITE & ASSOC. ELECTRO/MECHANICAL ENGINEERS ATTN RICHARD P. WHITE 77 PELHAM ISLE ROAD SUDSBURY, MA 01776

ROCKWELL INTERNATIONAL CORPORATION
COLUMBUS AIRCRAFT DIVISION,
PO BOX 1259
ATTN MARVIN SCHWEIGER
ATTN LOUIS BIAFORE
4300 E. 5TH AVENUE
COLUMBUS, OH 43216

SANDIA LABORATORIES
ATTN WILLIAM R. LEUENBERGER, DIV 2323
ATTN JERRY HOOD
ATTN NED KELTNER
ATTN ANTHONY VENERUSO, DIV 4742
ALBUQUERQUE, NM 87185

SCIENCE APPLICATIONS, INC ATTN J. ISEMAN 8400 WESTPARK DR MCLEAN, VA 22102

SIKORSKY AIRCRAFT ATTN J. R. SOEHNLEIN NORTH MAIN STREET STRATFORD, CT 06602

STEIN ENGINEERING SERVICES, INC 5602 E. MONTEROSA PHOENIX, AZ 85018

TRANS-TECH, INC ATTN L. DOMINGUES 12 MEEM AVE GAITHERSBURG, MD 20760

TRITEC, INC ATTN L. SIERACKI PO BOX 56 COLUMBIA, MD 21045

UNITED TECHNOLOGIES RESEARCH CENTER
ATTN R. E. OLSON, MGR FLUID
DYNAMICS LABORATORY
400 MAIN STREET
E. HARTFORD, CT 06108

VOUGHT CORPORATION PO BOX 225907 ATTN KELLEY FLING DALLAS, TX 75265

US ARMY ELECTRONICS RESEARCH
6 DEVELOPMENT COMMAND
ATTN TECHNICAL DIRECTOR, DRDEL-CT

HARRY DIAMOND LABORATORIES
ATTN CO/TD/TSO/DIVISION DIRECTORS
ATTN RECORD COPY, 81200
ATTN HDL LIBRARY, (2 COPIES), 81100
ATTN HDL LIBRARY, (WOODBRIDGE), 81100
ATTN TECHNICAL REPORTS BRANCH, 81300
ATTN LEGAL OFFICE, 97000
ATTN CHAIRMAN, EDITORIAL COMMITTEE
ATTN CORRIGAN, J., 20240
ATTN CHIEF, 13000
ATTN CHIEF, 13400 (20 COPIES)

#